



دانشگاه سمنان

Semnan University
Faculty of Mechanical Engineering

دانشگاه مهندسی مکانیک



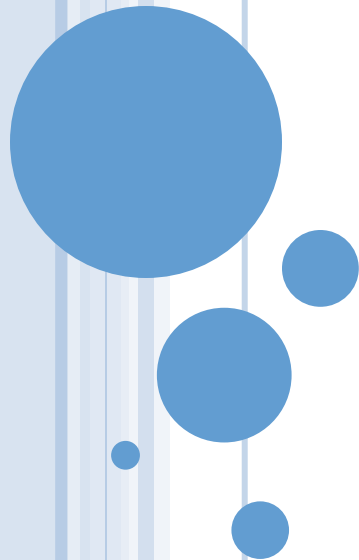
دانشگاه مهندسی مکانیک

درس رباتیک

ROBOTICS

Chapter 2 – Kinematics

Class Lecture



□ CONTENTS:

❖ Chapter 1: Introduction

→ ❖ Chapter 2: **Kinematics**

❖ Chapter 3: Differential Kinematics and Statics

❖ Chapter 4: Trajectory Planning

❖ Chapter 5: Actuators and Sensors

❖ Chapter 6: Control Architecture

2. KINEMATICS

- ❑ A manipulator:
 - ❖ Kinematic chain of rigid bodies (links) connected by means of revolute or prismatic joints.

- ❑ The derivation of the **Direct Kinematics Equation** allows the end-effector position and orientation (pose) to be expressed as a function of the joint variables.

- ❑ With reference to a minimal representation of orientation, the concept of **Operational Space** is introduced and its relationship with the **Joint Space** is established.



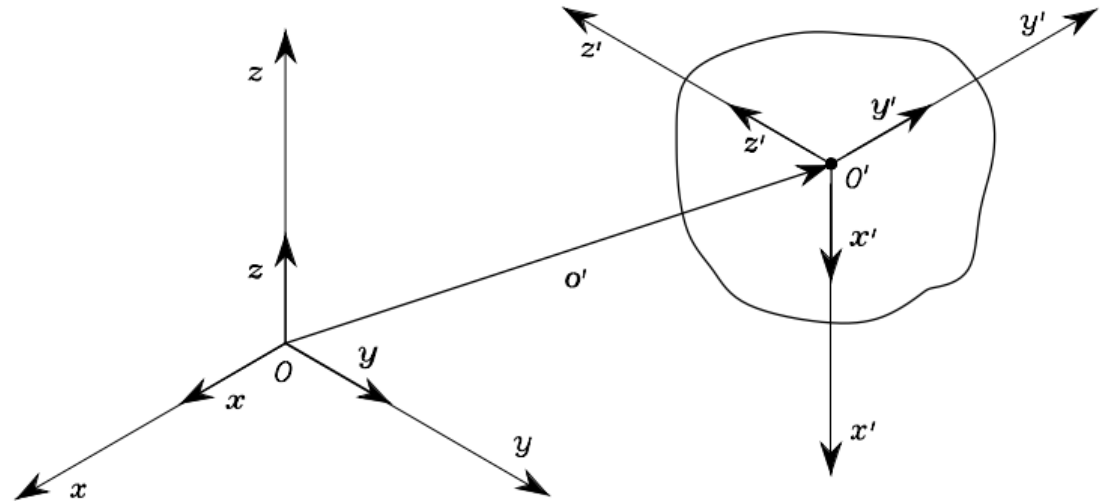
2.1 POSE OF A RIGID BODY

- A rigid body is completely described in space by its **position** and **orientation** (in brief pose) with respect to a reference frame.

$$\mathbf{o}' = o'_x \mathbf{x} + o'_y \mathbf{y} + o'_z \mathbf{z},$$

- Components of the vector along the frame axes

$$\mathbf{o}' = \begin{bmatrix} o'_x \\ o'_y \\ o'_z \end{bmatrix}$$



2.1 POSE OF A RIGID BODY

- O–xyz: Reference frame
- O'–x'y'z': Orthonormal frame attached to the body and express its unit vectors with respect to the reference frame.

$$\mathbf{x}' = x'_x \mathbf{x} + x'_y \mathbf{y} + x'_z \mathbf{z}$$

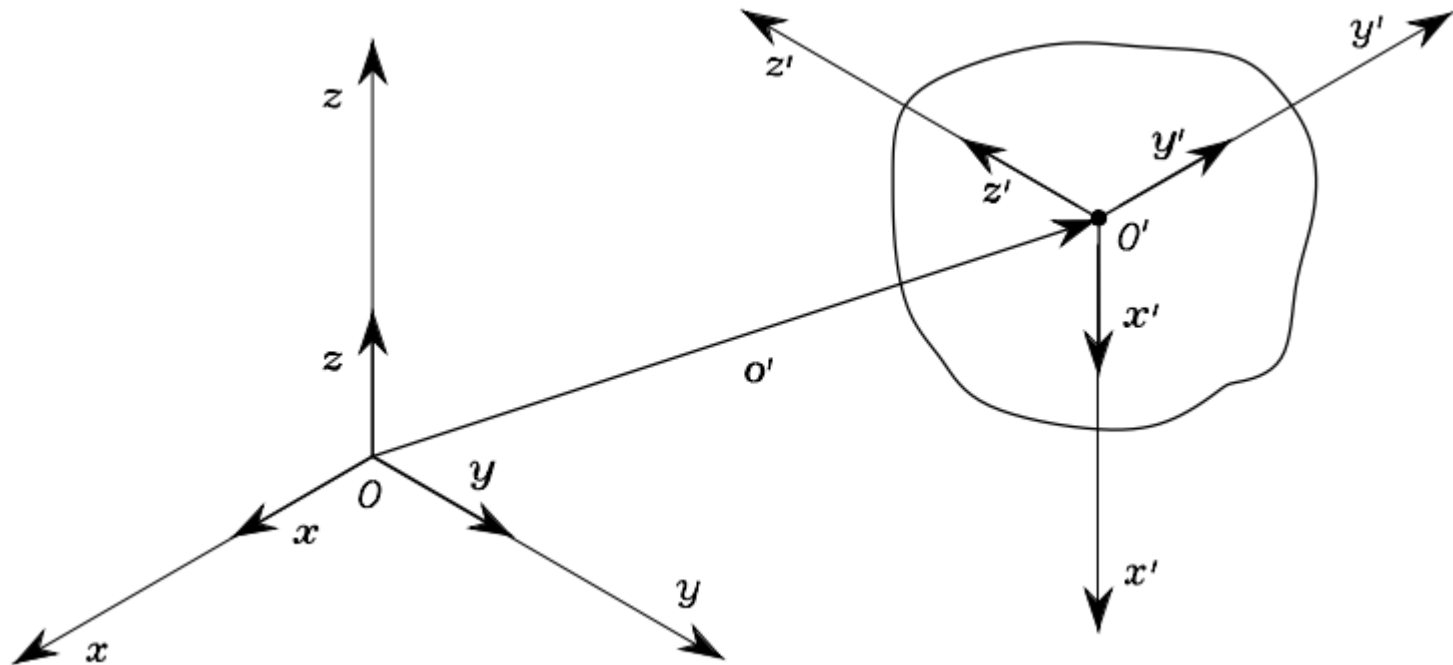
$$\mathbf{y}' = y'_x \mathbf{x} + y'_y \mathbf{y} + y'_z \mathbf{z}$$

$$\mathbf{z}' = z'_x \mathbf{x} + z'_y \mathbf{y} + z'_z \mathbf{z}.$$

- ❖ The components of each unit vector are the direction cosines of the axes of frame O'–x'y'z' with respect to the reference frame O–xyz.

2.2 ROTATION MATRIX

- $O-xyz$ and $O'-x'y'z'$ frames



2.2 ROTATION MATRIX

- Unit vectors describing body orientation with respect to reference frame

$$\mathbf{R} = \begin{bmatrix} \mathbf{x}' & \mathbf{y}' & \mathbf{z}' \end{bmatrix} = \begin{bmatrix} x'_x & y'_x & z'_x \\ x'_y & y'_y & z'_y \\ x'_z & y'_z & z'_z \end{bmatrix} = \begin{bmatrix} \mathbf{x}'^T \mathbf{x} & \mathbf{y}'^T \mathbf{x} & \mathbf{z}'^T \mathbf{x} \\ \mathbf{x}'^T \mathbf{y} & \mathbf{y}'^T \mathbf{y} & \mathbf{z}'^T \mathbf{y} \\ \mathbf{x}'^T \mathbf{z} & \mathbf{y}'^T \mathbf{z} & \mathbf{z}'^T \mathbf{z} \end{bmatrix}$$

- Column vectors of matrix \mathbf{R} are **mutually orthogonal** since they represent the unit vectors of an orthonormal frame

$$\mathbf{x}'^T \mathbf{y}' = 0 \quad \mathbf{y}'^T \mathbf{z}' = 0 \quad \mathbf{z}'^T \mathbf{x}' = 0.$$

- Also, they have unit norm

$$\mathbf{x}'^T \mathbf{x}' = 1 \quad \mathbf{y}'^T \mathbf{y}' = 1 \quad \mathbf{z}'^T \mathbf{z}' = 1.$$



2.2 ROTATION MATRIX

$$R = \begin{bmatrix} \mathbf{x}' & \mathbf{y}' & \mathbf{z}' \end{bmatrix} = \begin{bmatrix} x'_x & y'_x & z'_x \\ x'_y & y'_y & z'_y \\ x'_z & y'_z & z'_z \end{bmatrix} = \begin{bmatrix} \mathbf{x}'^T \mathbf{x} & \mathbf{y}'^T \mathbf{x} & \mathbf{z}'^T \mathbf{x} \\ \mathbf{x}'^T \mathbf{y} & \mathbf{y}'^T \mathbf{y} & \mathbf{z}'^T \mathbf{y} \\ \mathbf{x}'^T \mathbf{z} & \mathbf{y}'^T \mathbf{z} & \mathbf{z}'^T \mathbf{z} \end{bmatrix}$$

- As a consequence, R is an orthogonal matrix

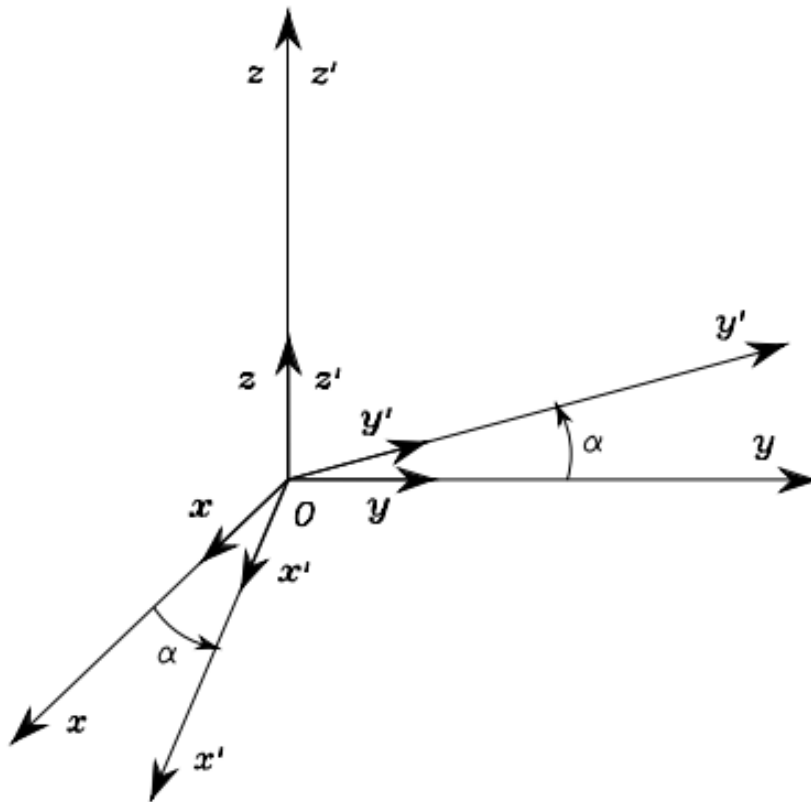
$$R^T R = I_3$$

$$\rightarrow R^T = R^{-1}$$

- ❖ Right-handed frame: $\det(R) = 1$
- ❖ Left-handed frame: $\det(R) = -1$

2.2.1 ELEMENTARY ROTATIONS

- Elementary rotations of the reference frame about one of the coordinate axes
 - ❖ Reference frame O-xyz is rotated by an angle α about axis z



$$x' = \begin{bmatrix} \cos \alpha \\ \sin \alpha \\ 0 \end{bmatrix}$$

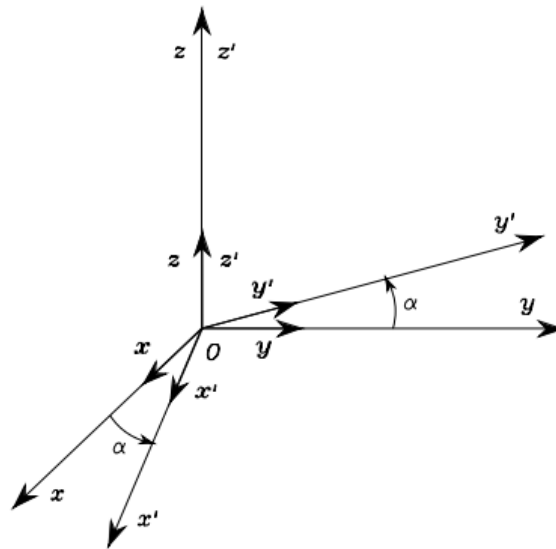
$$y' = \begin{bmatrix} -\sin \alpha \\ \cos \alpha \\ 0 \end{bmatrix}$$

$$z' = \begin{bmatrix} 0 \\ 0 \\ 1 \end{bmatrix}$$

2.2.1 ELEMENTARY ROTATIONS

- Rotation matrix of frame O–x'y'z' with respect to frame O-xyz is

$$R_z(\alpha) = \begin{bmatrix} \cos \alpha & -\sin \alpha & 0 \\ \sin \alpha & \cos \alpha & 0 \\ 0 & 0 & 1 \end{bmatrix}$$



2.2.1 ELEMENTARY ROTATIONS

- Rotations by an angle β about axis y

$$\mathbf{R}_y(\beta) = \begin{bmatrix} \cos \beta & 0 & \sin \beta \\ 0 & 1 & 0 \\ -\sin \beta & 0 & \cos \beta \end{bmatrix}$$

- Rotation by an angle γ about axis x

$$\mathbf{R}_x(\gamma) = \begin{bmatrix} 1 & 0 & 0 \\ 0 & \cos \gamma & -\sin \gamma \\ 0 & \sin \gamma & \cos \gamma \end{bmatrix}$$

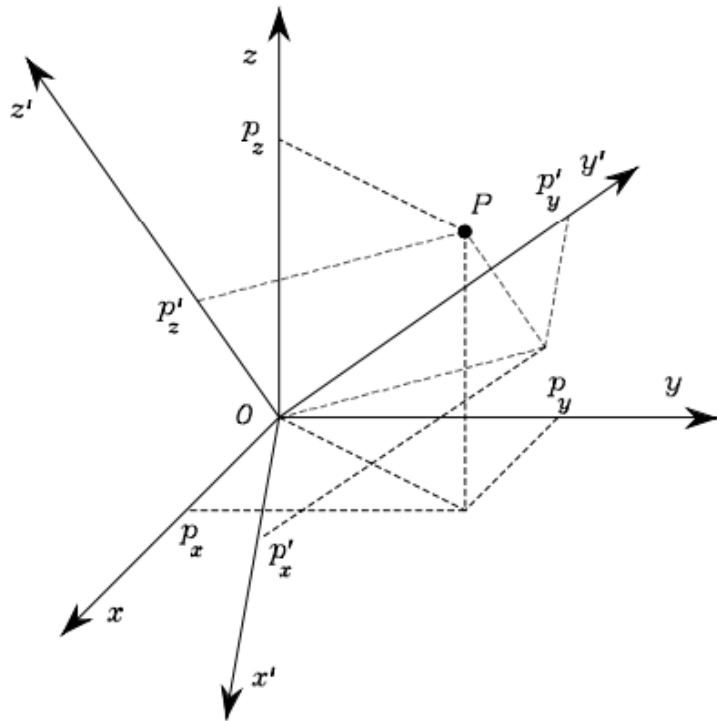
❖ Also:

$$\rightarrow \mathbf{R}_k(-\vartheta) = \mathbf{R}_k^T(\vartheta) \quad k = x, y, z.$$



2.2.2 REPRESENTATION OF A VECTOR

- With coincident origins



$$\mathbf{p} = \begin{bmatrix} p_x \\ p_y \\ p_z \end{bmatrix} \quad \mathbf{p}' = \begin{bmatrix} p'_x \\ p'_y \\ p'_z \end{bmatrix}$$

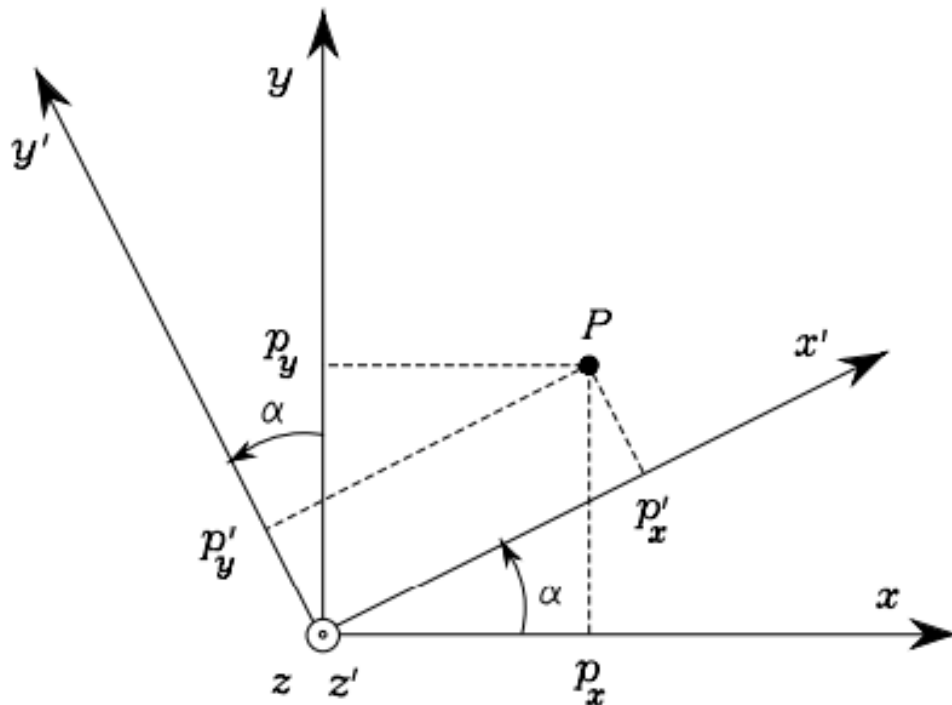
$$\mathbf{p} = p'_x \mathbf{x}' + p'_y \mathbf{y}' + p'_z \mathbf{z}' = \begin{bmatrix} \mathbf{x}' & \mathbf{y}' & \mathbf{z}' \end{bmatrix} \mathbf{p}'$$

$$\Rightarrow \mathbf{p} = \mathbf{R} \mathbf{p}'$$

$$\Rightarrow \mathbf{p}' = \mathbf{R}^T \mathbf{p}$$

2.2.2 REPRESENTATION OF A VECTOR

□ Example 2.1



$$p_x = p'_x \cos \alpha - p'_y \sin \alpha$$

$$p_y = p'_x \sin \alpha + p'_y \cos \alpha$$

$$p_z = p'_z.$$

2.2.3 ROTATION OF A VECTOR

- A rotation matrix can be also interpreted as the **matrix operator** allowing rotation of a vector by a given angle about an arbitrary axis in space.

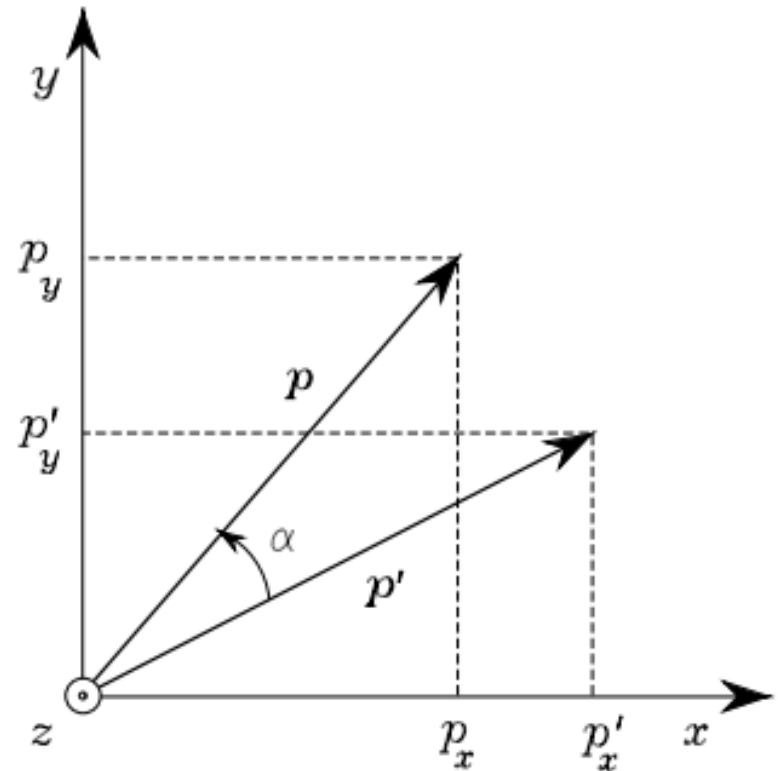
- Example 2.2

$$p_x = p'_x \cos \alpha - p'_y \sin \alpha$$

$$p_y = p'_x \sin \alpha + p'_y \cos \alpha$$

$$p_z = p'_z$$

$$\rightarrow \mathbf{p} = \mathbf{R}_z(\alpha)\mathbf{p}'$$



2.2.3 ROTATION OF A VECTOR

- A rotation matrix attains three equivalent geometrical meanings:
 - ❖ Mutual orientation between two coordinate frames
(its column vectors are the direction cosines of the axes of the rotated frame with respect to the original frame)
 - ❖ Coordinate transformation between the coordinates of a point expressed in two different frames (with common origin)
 - ❖ An operator that allows the rotation of a vector in the same coordinate frame.



2.3 COMPOSITION OF ROTATION MATRICES

- $O-x_0y_0z_0$, $O-x_1y_1z_1$, $O-x_2y_2z_2$ (three frames with common origin O)
- The vector p : position of a generic point in space
 - ❖ p^0, p^1, p^2 : the expressions of p in the three frames.

$$\begin{aligned}
 p^1 &= R_2^1 p^2 \\
 p^0 &= R_1^0 p^1 \quad \longrightarrow \quad R_2^0 = R_1^0 R_2^1 \\
 p^0 &= R_2^0 p^2
 \end{aligned}$$

- ❖ The overall rotation can be expressed as a sequence of partial rotations

- Also:

$$\longrightarrow R_i^j = (R_j^i)^{-1} = (R_j^i)^T$$



2.3 COMPOSITION OF ROTATION MATRICES

- ❑ The frame with respect to which the rotation occurs is termed **current frame**.
- ❑ Composition of successive rotations is then obtained by **postmultiplication** of the rotation matrices following the given order of rotations



2.3 COMPOSITION OF ROTATION MATRICES

- ❑ Successive rotations can be also specified by constantly referring them to the **initial frame**.
- ❑ In this case, the rotations are made with respect to a fixed frame.

$$\bar{R}_2^0 = R_1^0 R_0^1 \bar{R}_2^1 R_1^0 \quad \rightarrow \quad \bar{R}_2^0 = \bar{R}_2^1 R_1^0$$

- ❑ Hence, it can be stated that composition of successive rotations with respect to a fixed frame is obtained by **premultiplication** of the single rotation matrices in the order of the given sequence of rotations.

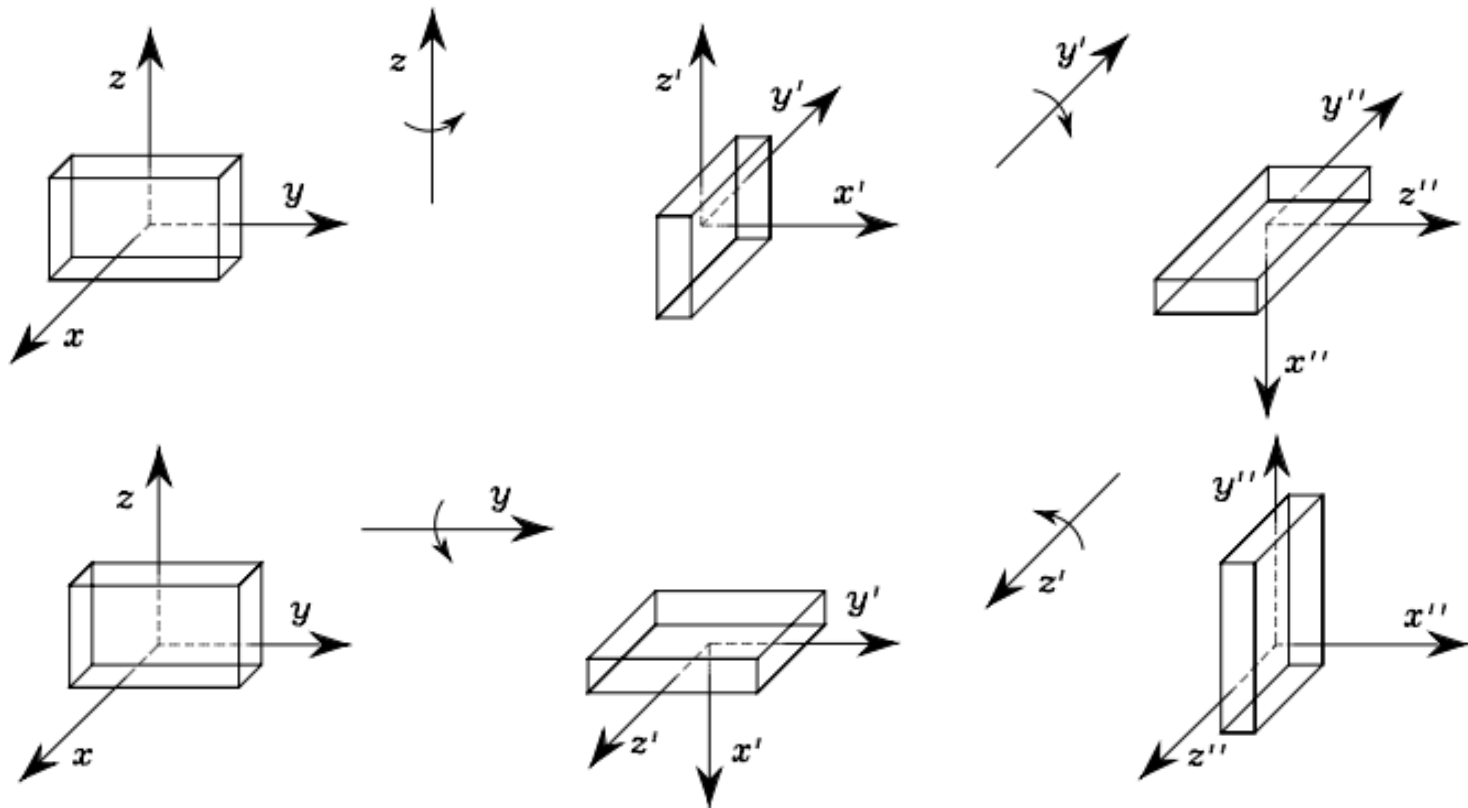
❖ An important issue of composition of rotations is that the matrix product is not commutative.



2.3 COMPOSITION OF ROTATION MATRICES

□ Example 2.3

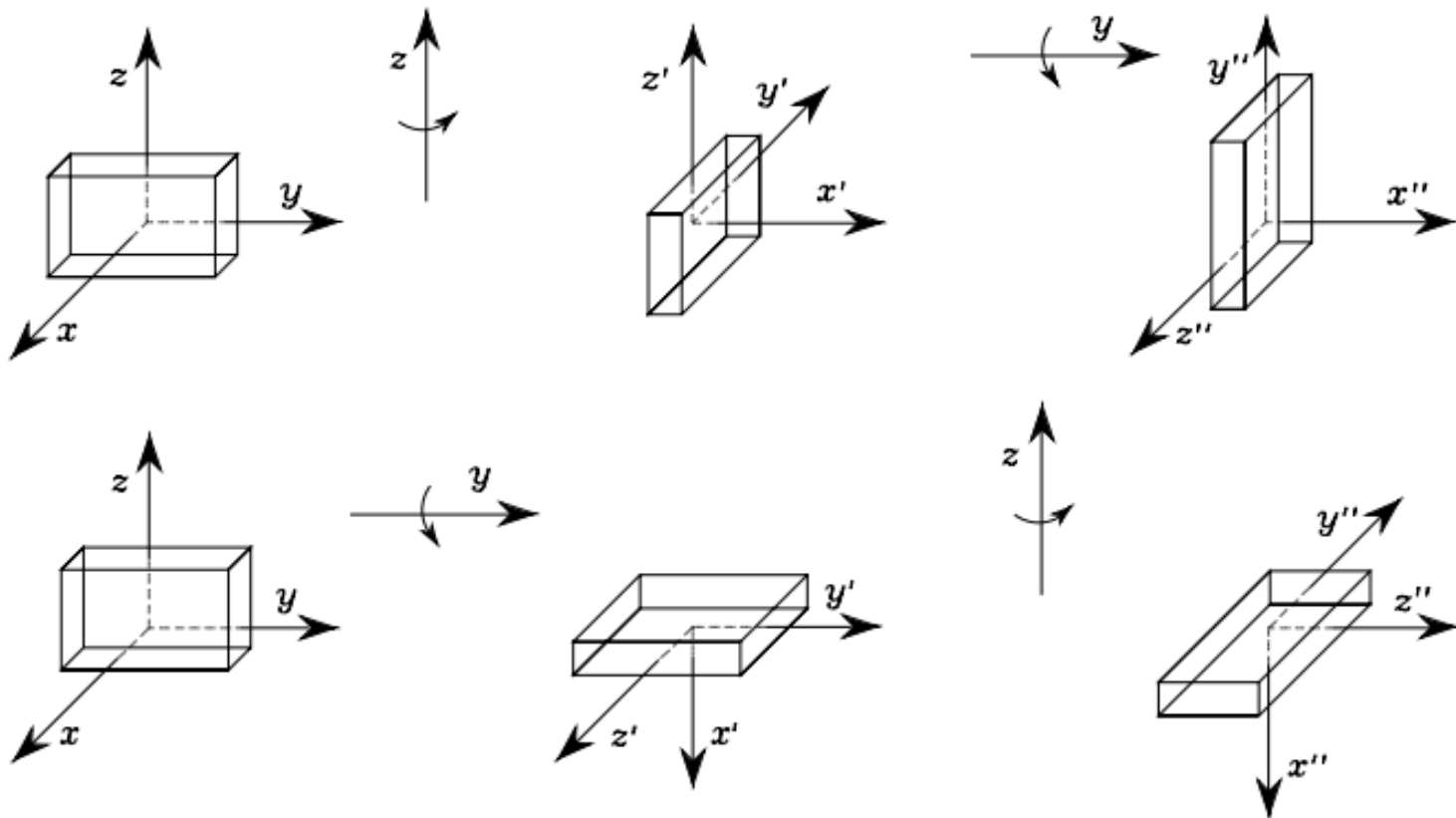
❖ Successive rotations of an object about axes of current frame



2.3 COMPOSITION OF ROTATION MATRICES

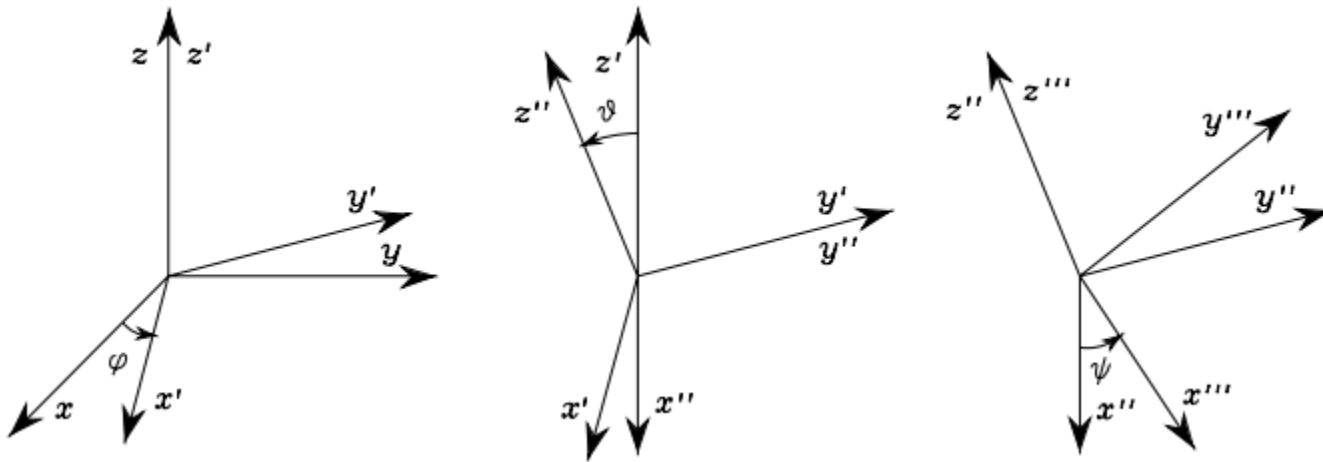
□ Example 2.3

❖ Successive rotations of an object about axes of fixed frame



2.4 EULER ANGLES

- A minimal representation of orientation can be obtained by using a set of three angles $[\phi \ \vartheta \ \psi]^T$.
- 2.4.1 ZYZ Angles
 - ❖ Rotate the reference frame by the angle ϕ about axis z
 - ❖ Rotate the current frame by the angle ϑ about axis y
 - ❖ Rotate the current frame by the angle ψ about axis z



2.4 EULER ANGLES

□ 2.4.1 ZYZ Angles

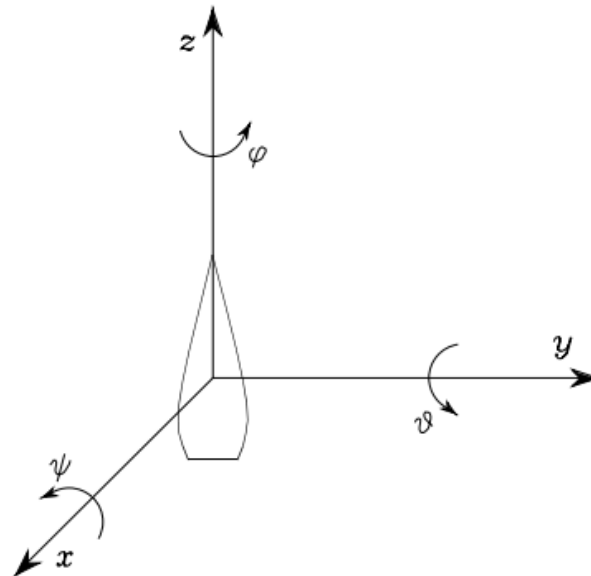
- ❖ The resulting frame orientation is obtained by composition of rotations with respect to current frames

$$\begin{aligned}
 \mathbf{R}(\phi) &= \mathbf{R}_z(\varphi)\mathbf{R}_{y'}(\vartheta)\mathbf{R}_{z''}(\psi) \\
 &= \begin{bmatrix} c_\varphi c_\vartheta c_\psi - s_\varphi s_\psi & -c_\varphi c_\vartheta s_\psi - s_\varphi c_\psi & c_\varphi s_\vartheta \\ s_\varphi c_\vartheta c_\psi + c_\varphi s_\psi & -s_\varphi c_\vartheta s_\psi + c_\varphi c_\psi & s_\varphi s_\vartheta \\ -s_\vartheta c_\psi & s_\vartheta s_\psi & c_\vartheta \end{bmatrix}
 \end{aligned}$$

2.4 EULER ANGLES

□ 2.4.2 RPY Angles

- ❖ Representation of orientation in the aeronautical field.
- ❖ These are the ZYX angles, also called Roll–Pitch–Yaw angles, to denote the typical changes of attitude of an aircraft.
- ❖ The angles $[\phi \ \theta \ \psi]^T$ represent rotations defined with respect to a fixed frame attached to the center of mass of the aircraft.



2.4 EULER ANGLES

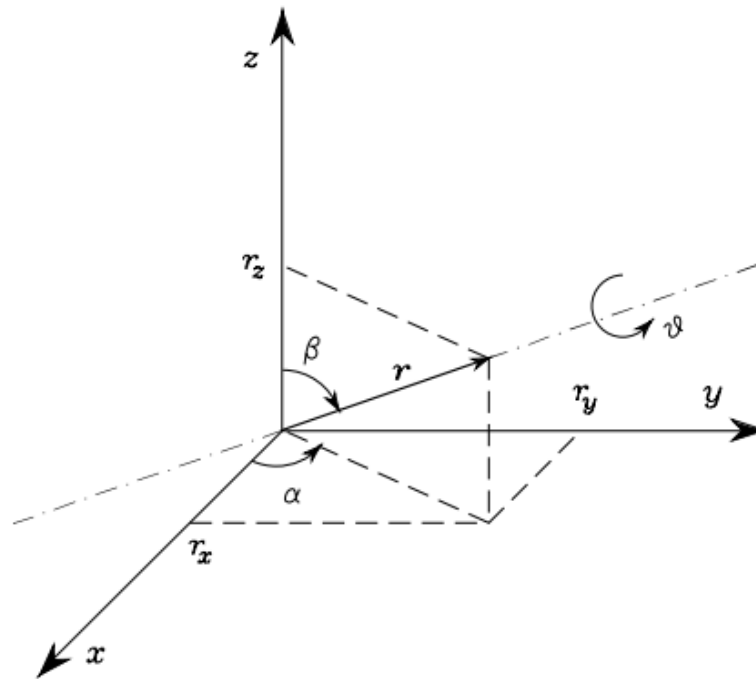
□ 2.4.2 RPY Angles

- ❖ Rotate the reference frame by the angle ψ about axis x (yaw)
- ❖ Rotate the reference frame by the angle ϑ about axis y (pitch)
- ❖ Rotate the reference frame by the angle ϕ about axis z (roll)

$$\begin{aligned}
 \mathbf{R}(\phi) &= \mathbf{R}_z(\phi)\mathbf{R}_y(\vartheta)\mathbf{R}_x(\psi) \\
 &= \begin{bmatrix} c_\phi c_\vartheta & c_\phi s_\vartheta s_\psi - s_\phi c_\psi & c_\phi s_\vartheta c_\psi + s_\phi s_\psi \\ s_\phi c_\vartheta & s_\phi s_\vartheta s_\psi + c_\phi c_\psi & s_\phi s_\vartheta c_\psi - c_\phi s_\psi \\ -s_\vartheta & c_\vartheta s_\psi & c_\vartheta c_\psi \end{bmatrix}
 \end{aligned}$$

2.5 ANGLE AND AXIS

- A nonminimal representation: rotation of a given angle about an axis in space (with 4 parameters)
- This can be advantageous in the problem of trajectory planning for a manipulator's end-effector orientation.



2.5 ANGLE AND AXIS

- Let $\mathbf{r} = [r_x \ r_y \ r_z]^T$ be the unit vector of a rotation axis with respect to the reference frame $O-xyz$.
- In order to derive the rotation matrix $R(\vartheta, \mathbf{r})$ expressing the rotation of an angle ϑ about axis \mathbf{r}

$$R(\vartheta, \mathbf{r}) = R_z(\alpha)R_y(\beta)R_z(\vartheta)R_y(-\beta)R_z(-\alpha)$$

$$\sin \alpha = \frac{r_y}{\sqrt{r_x^2 + r_y^2}} \quad \cos \alpha = \frac{r_x}{\sqrt{r_x^2 + r_y^2}}$$

$$\sin \beta = \frac{r_z}{\sqrt{r_x^2 + r_y^2}} \quad \cos \beta = r_z.$$

$$\rightarrow R(\vartheta, \mathbf{r}) = \begin{bmatrix} r_x^2(1 - c_\vartheta) + c_\vartheta & r_x r_y(1 - c_\vartheta) - r_z s_\vartheta & r_x r_z(1 - c_\vartheta) + r_y s_\vartheta \\ r_x r_y(1 - c_\vartheta) + r_z s_\vartheta & r_y^2(1 - c_\vartheta) + c_\vartheta & r_y r_z(1 - c_\vartheta) - r_x s_\vartheta \\ r_x r_z(1 - c_\vartheta) - r_y s_\vartheta & r_y r_z(1 - c_\vartheta) + r_x s_\vartheta & r_z^2(1 - c_\vartheta) + c_\vartheta \end{bmatrix}$$



2.5 ANGLE AND AXIS

- The inverse problem

$$\mathbf{R} = \begin{bmatrix} r_{11} & r_{12} & r_{13} \\ r_{21} & r_{22} & r_{23} \\ r_{31} & r_{32} & r_{33} \end{bmatrix} \quad \rightarrow \quad \vartheta = \cos^{-1} \left(\frac{r_{11} + r_{22} + r_{33} - 1}{2} \right)$$

$$\mathbf{r} = \frac{1}{2 \sin \vartheta} \begin{bmatrix} r_{32} - r_{23} \\ r_{13} - r_{31} \\ r_{21} - r_{12} \end{bmatrix},$$

- ❖ The three components of \mathbf{r} are not independent but are constrained:

$$r_x^2 + r_y^2 + r_z^2 = 1$$

2.6 UNIT QUATERNION

- The drawbacks of the angle/axis representation can be overcome by a different four-parameter representation; namely, the **Unit Quaternion**

$$Q = \{\eta, \epsilon\}$$

$$\eta = \cos \frac{\vartheta}{2}$$



$$\epsilon = \sin \frac{\vartheta}{2} r$$

2.6 UNIT QUATERNION

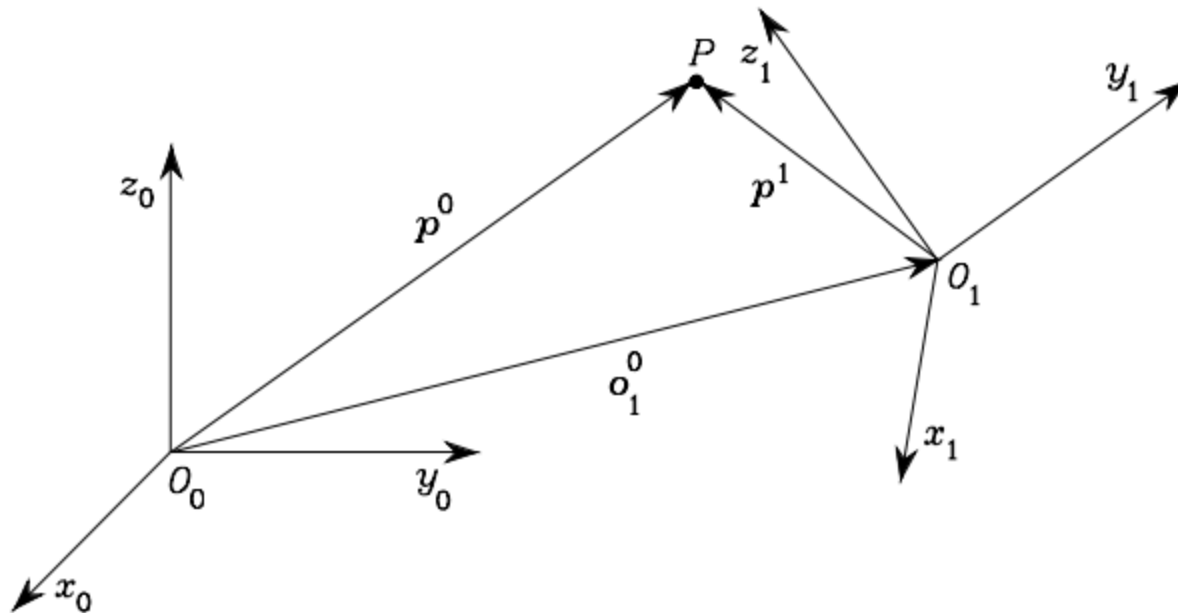
- η : the scalar part of the quaternion while
- $\boldsymbol{\epsilon} = [\epsilon_x \ \epsilon_y \ \epsilon_z]^T$: the vector part of the quaternion.

$$\eta^2 + \epsilon_x^2 + \epsilon_y^2 + \epsilon_z^2 = 1$$

$$\longrightarrow \mathbf{R}(\eta, \boldsymbol{\epsilon}) = \begin{bmatrix} 2(\eta^2 + \epsilon_x^2) - 1 & 2(\epsilon_x\epsilon_y - \eta\epsilon_z) & 2(\epsilon_x\epsilon_z + \eta\epsilon_y) \\ 2(\epsilon_x\epsilon_y + \eta\epsilon_z) & 2(\eta^2 + \epsilon_y^2) - 1 & 2(\epsilon_y\epsilon_z - \eta\epsilon_x) \\ 2(\epsilon_x\epsilon_z - \eta\epsilon_y) & 2(\epsilon_y\epsilon_z + \eta\epsilon_x) & 2(\eta^2 + \epsilon_z^2) - 1 \end{bmatrix}$$

2.7 HOMOGENEOUS TRANSFORMATIONS

- The position of a rigid body in space:
 - ❖ Position of a point on the body with respect to a reference frame (translation)
 - ❖ Components of the unit vectors (orientation) of a frame attached to the body with respect to the same reference frame (rotation)



2.7 HOMOGENEOUS TRANSFORMATIONS

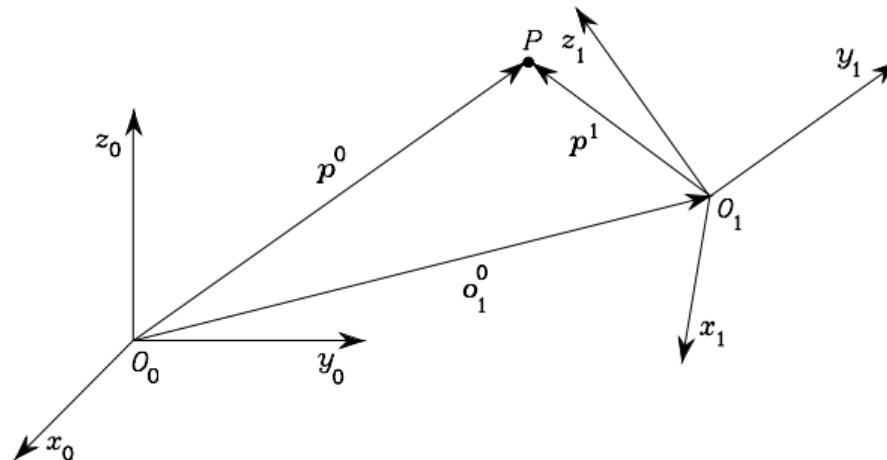
- Coordinate transformation (translation + rotation) of a bound vector between two frames:

- ✓ R_0^1 : Rotation matrix of Frame 1 with respect to Frame 0

$$p^0 = o_1^0 + R_1^0 p^1$$

- The inverse transformation:

$$p^1 = -R_1^{0T} o_1^0 + R_1^{0T} p^0 \quad \longrightarrow \quad p^1 = -R_0^1 o_1^0 + R_0^1 p^0$$



2.7 HOMOGENEOUS TRANSFORMATIONS

□ The **Homogeneous Representation** of a generic vector $\mathbf{p} : (\tilde{\mathbf{p}})$

- ❖ In order to achieve a compact representation of the relationship between the coordinates of the same point in two different frames

$$\tilde{\mathbf{p}} = \begin{bmatrix} \mathbf{p} \\ 1 \end{bmatrix} \longrightarrow \begin{aligned} \tilde{\mathbf{p}}^0 &= \mathbf{A}_1^0 \tilde{\mathbf{p}}^1 \\ \tilde{\mathbf{p}}^1 &= \mathbf{A}_0^1 \tilde{\mathbf{p}}^0 = (\mathbf{A}_1^0)^{-1} \tilde{\mathbf{p}}^0 \end{aligned}$$

- ❖ *Homogeneous Transformation Matrix*

$$\mathbf{A}_1^0 = \begin{bmatrix} \mathbf{R}_1^0 & \mathbf{o}_1^0 \\ \mathbf{0}^T & 1 \end{bmatrix}$$

2.7 HOMOGENEOUS TRANSFORMATIONS

□ *Homogeneous Transformation Matrix*

$$A_1^0 = \begin{bmatrix} R_1^0 & o_1^0 \\ \mathbf{0}^T & 1 \end{bmatrix}$$

$$A_0^1 = \begin{bmatrix} R_1^{0T} & -R_1^{0T} o_1^0 \\ \mathbf{0}^T & 1 \end{bmatrix} = \begin{bmatrix} R_0^1 & -R_0^1 o_1^0 \\ \mathbf{0}^T & 1 \end{bmatrix}$$

❖ Notice that:

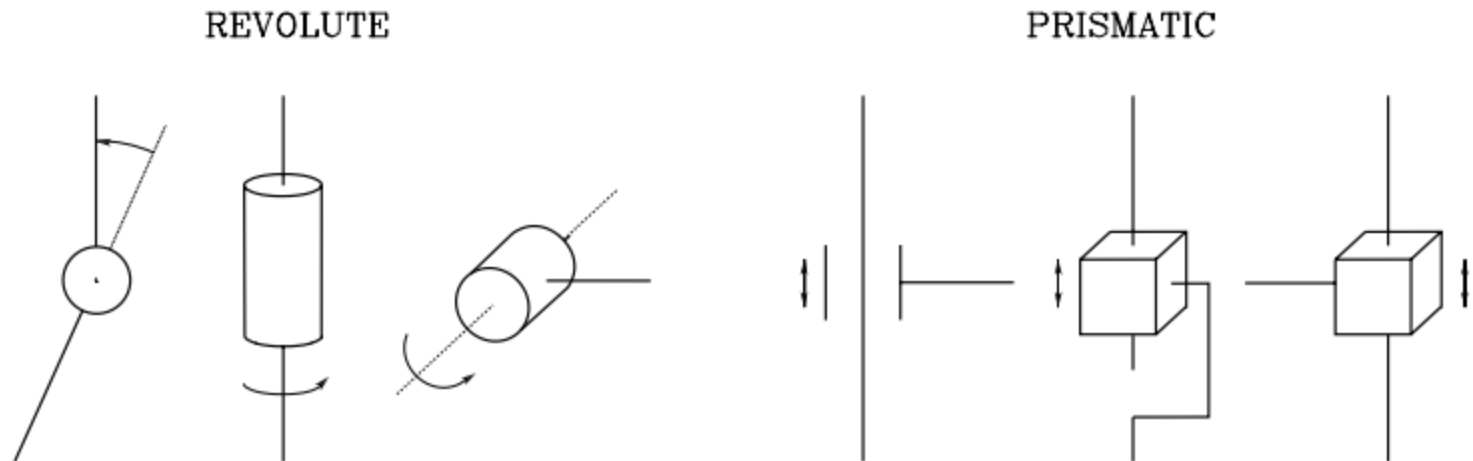
$$A^{-1} \neq A^T$$

$$\tilde{p}^0 = A_1^0 A_2^1 \dots A_n^{n-1} \tilde{p}^n$$



2.8 DIRECT KINEMATICS

- A manipulator:
 - ❖ Series of rigid bodies (links) connected by means of kinematic pairs or joints
- Joints:
 - ❖ Revolute
 - ❖ Prismatic



2.8 DIRECT KINEMATICS

- ❖ The whole structure forms a Kinematic Chain.
 - ✓ One end of the chain is constrained to a base.
 - ✓ The other end is an end-effector (gripper, tool)
- ❖ Open kinematic chain (only one sequence of links connecting the two ends)
- ❖ Closed kinematic chain (a sequence of links forms a loop)
- ❖ Characterized by a number of degrees of freedom (DOFs)
 - ✓ Uniquely determine its posture.
 - ✓ Each DOF is typically associated with a joint articulation and constitutes a joint variable

❑ Direct kinematics:

- ❖ Compute the pose of the end-effector as a function of the joint variables

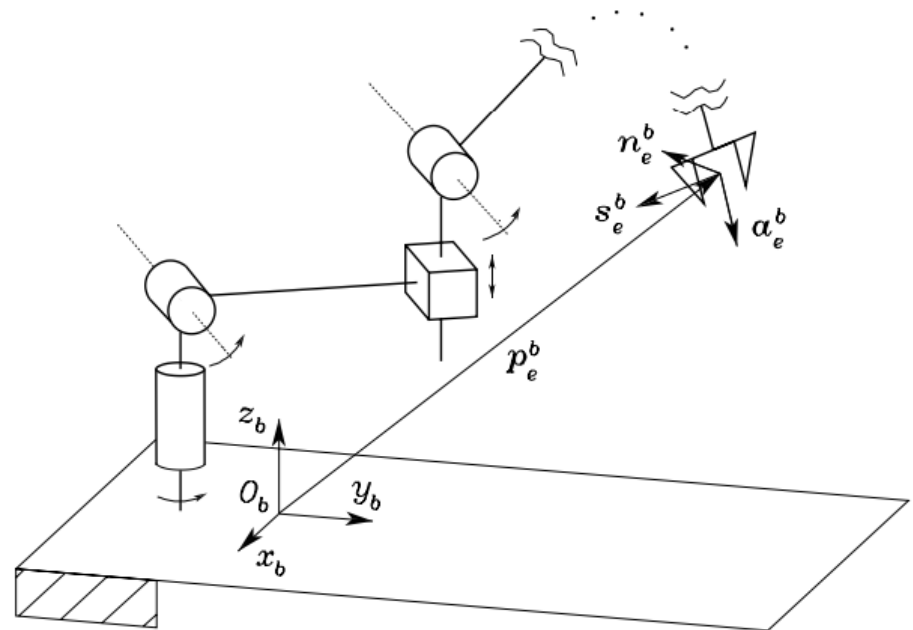


2.8 DIRECT KINEMATICS

- Direct kinematics function homogeneous transformation matrix

$$T_e^b(q) = \begin{bmatrix} n_e^b(q) & s_e^b(q) & a_e^b(q) & p_e^b(q) \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

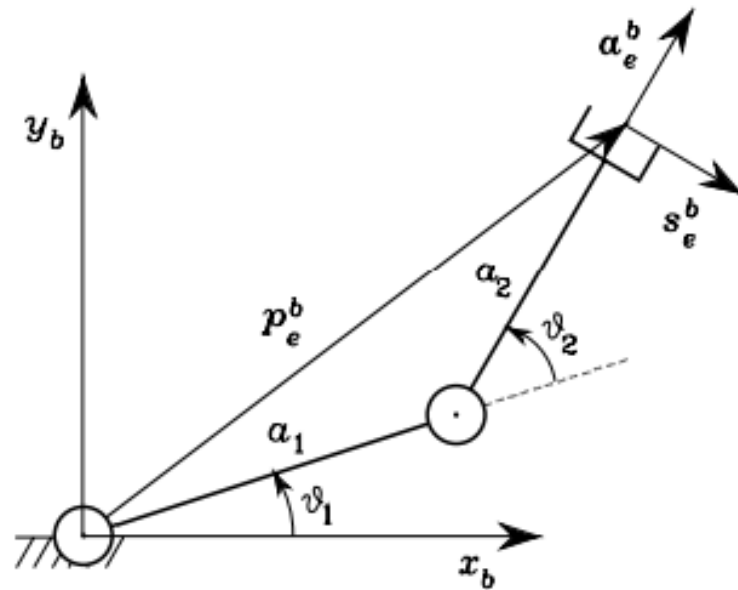
❖ n_e , s_e , a_e and p_e are a function of q



2.8 DIRECT KINEMATICS

□ Example 2.4

❖ Two-link planar arm



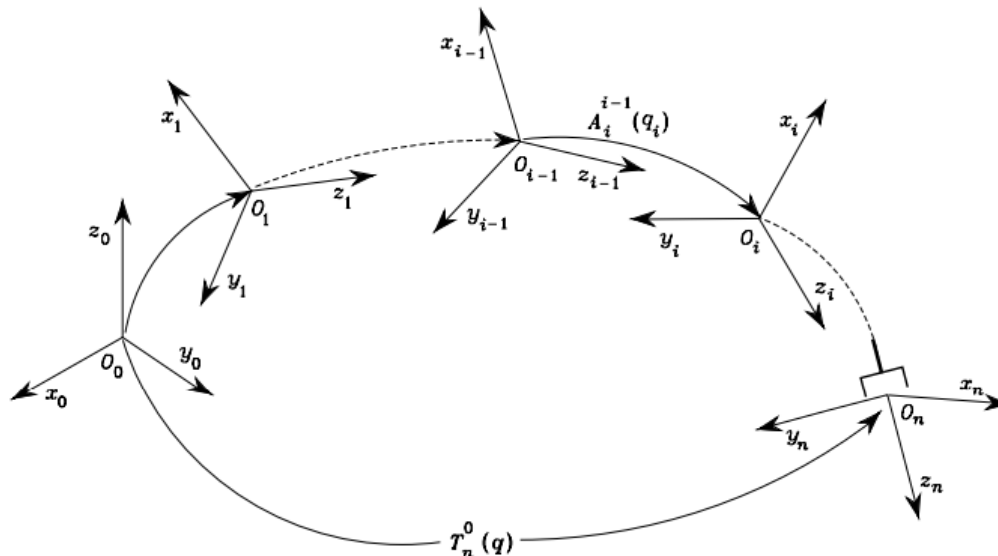
$$T_e^b(\mathbf{q}) = \begin{bmatrix} \mathbf{n}_e^b & \mathbf{s}_e^b & \mathbf{a}_e^b & \mathbf{p}_e^b \\ 0 & 0 & 0 & 1 \end{bmatrix} = \begin{bmatrix} 0 & s_{12} & c_{12} & a_1 c_1 + a_2 c_{12} \\ 0 & -c_{12} & s_{12} & a_1 s_1 + a_2 s_{12} \\ 1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

2.8 DIRECT KINEMATICS

□ 2.8.1 Open Chain

- ❖ An open-chain manipulator constituted by $n + 1$ links connected by n joints
- ❖ Define a coordinate frame attached to each link, from Link 0 to Link n
- ❖ The coordinate transformation describing the position and orientation of Frame n with respect to Frame 0:

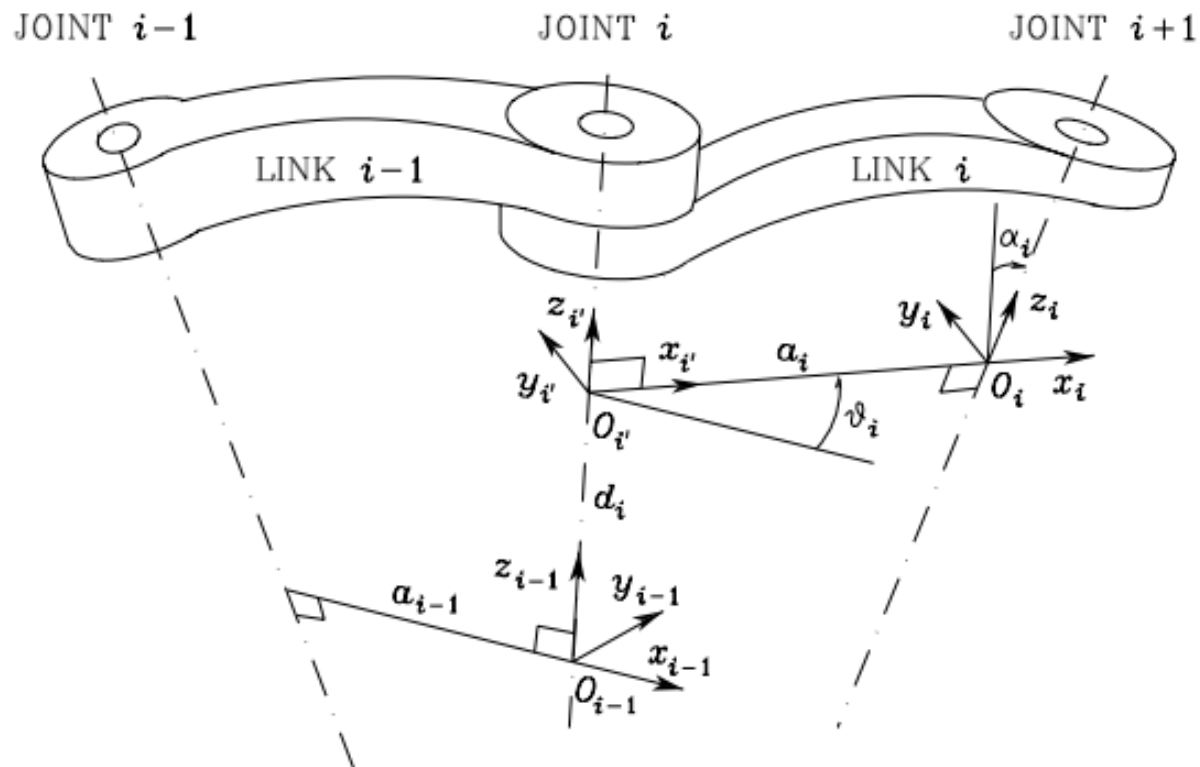
$$T_n^0(\mathbf{q}) = A_1^0(q_1)A_2^1(q_2) \dots A_n^{n-1}(q_n) \longrightarrow T_e^b(\mathbf{q}) = T_0^b T_n^0(\mathbf{q}) T_e^n$$



2.8 DIRECT KINEMATICS

□ 2.8.2 Denavit–Hartenberg Convention

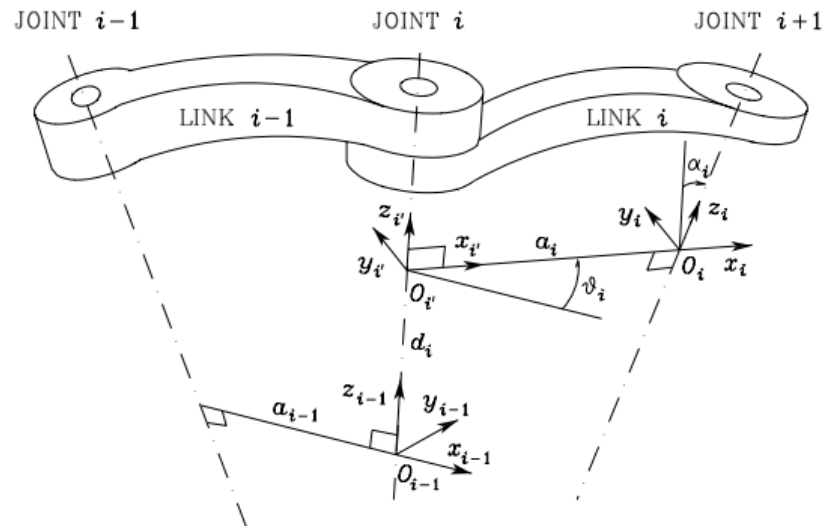
- ❖ A systematic, general method is to be derived to define the relative position and orientation of two consecutive links



2.8 DIRECT KINEMATICS

□ 2.8.2 Denavit–Hartenberg Convention

- ❖ Let Axis i denote the axis of the joint connecting Link $i - 1$ to Link i
- ❖ The Denavit–Hartenberg convention (DH) is adopted to define link Frame i :
 - ✓ Choose axis z_i along the axis of Joint $i + 1$
 - ✓ Locate the origin O_i and O_i'
 - ✓ Choose axis x_i along the common normal to axes z_{i-1} and z_i (from Joint i to Joint $i + 1$)
 - ✓ Choose axis y_i so as to complete a right-handed frame.



2.8 DIRECT KINEMATICS

□ 2.8.2 Denavit–Hartenberg Convention

❖ The Denavit–Hartenberg convention gives a nonunique definition of the link frame in the following cases:

- ✓ For Frame 0, only the direction of axis z_0 is specified; O_0 and x_0 can be arbitrarily chosen
- ✓ For Frame n (no Joint $n+1$) z_n is not uniquely defined while x_n has to be normal to axis z_{n-1}

(Typically, Joint n is revolute, and thus z_n is to be aligned with the direction of z_{n-1})

- ✓ When two consecutive axes are parallel, the common normal is not uniquely defined
- ✓ When two consecutive axes intersect, the direction of x_i is arbitrary
- ✓ When Joint i is prismatic, the direction of z_{i-1} is arbitrary

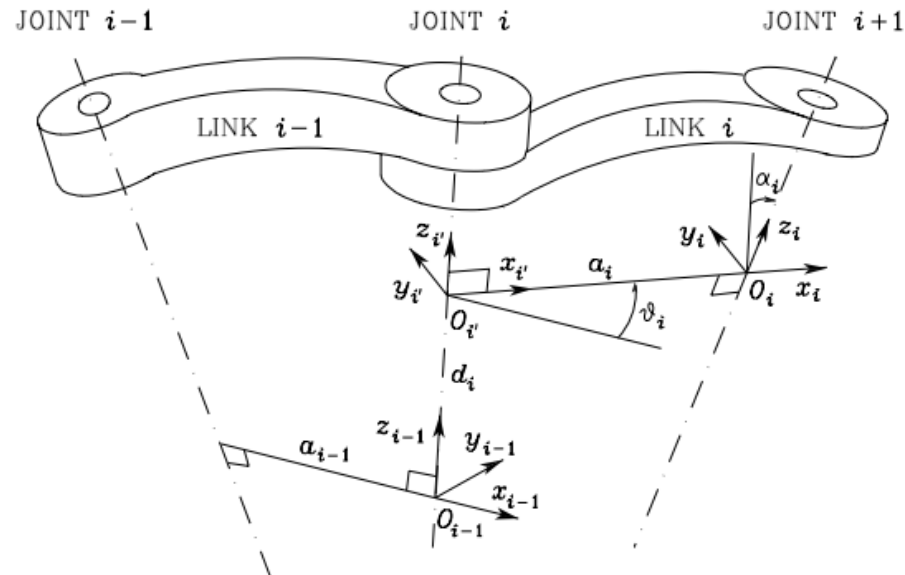


2.8 DIRECT KINEMATICS

□ 2.8.2 Denavit–Hartenberg Convention

❖ Parameters:

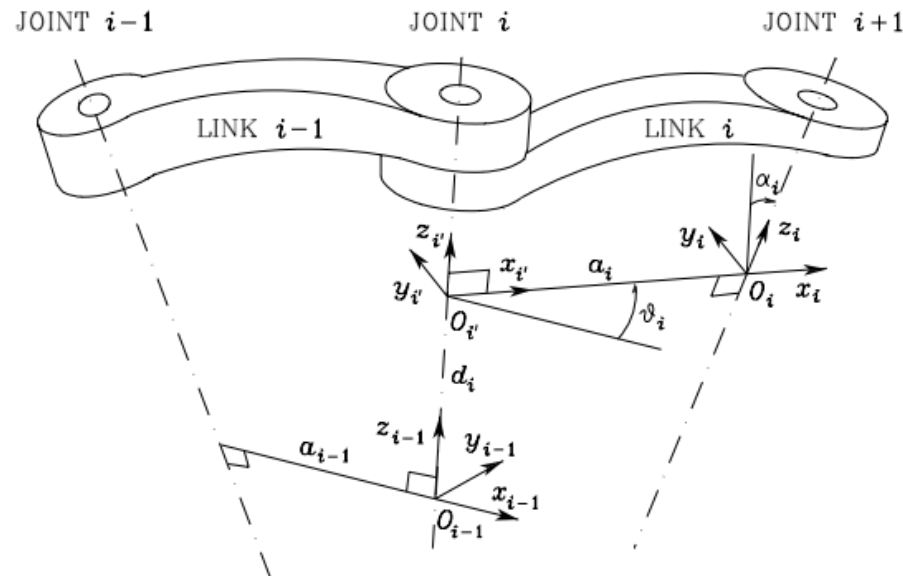
- ✓ a_i : Distance between O_i and O_i'
- ✓ d_i : Coordinate of O_i along z_{i-1}
- ✓ α_i : Angle between axes z_{i-1} and z_i about axis x_i (positive: counter-clockwise)
- ✓ θ_i : Angle between axes x_{i-1} and x_i about axis z_{i-1} (positive: counter-clockwise)



2.8 DIRECT KINEMATICS

□ 2.8.2 Denavit–Hartenberg Convention

- ❖ Two of the four parameters (a_i and α_i) are always constant and depend only on the geometry of connection between consecutive joints.
- ❖ *If Joint i is revolute the variable is θ_i*
- ❖ *If Joint i is prismatic the variable is d_i*



2.8 DIRECT KINEMATICS

□ 2.8.2 Denavit–Hartenberg Convention

❖ Coordinate transformation between Frame i and Frame $i - 1$:

1. Choose a frame aligned with Frame $i - 1$
2. Translate the chosen frame by d_i along axis z_{i-1} and rotate it by ϑ_i about axis z_{i-1}

$$A_{i'}^{i-1} = \begin{bmatrix} C\vartheta_i & -S\vartheta_i & 0 & 0 \\ S\vartheta_i & C\vartheta_i & 0 & 0 \\ 0 & 0 & 1 & d_i \\ 0 & 0 & 0 & 1 \end{bmatrix}$$



2.8 DIRECT KINEMATICS

□ 2.8.2 Denavit–Hartenberg Convention

❖ Coordinate transformation between Frame i and Frame $i - 1$:

3. Translate the frame aligned with Frame i' by a_i along x_i and rotate it by α_i about x_i

$$\mathbf{A}_i^{i'} = \begin{bmatrix} 1 & 0 & 0 & a_i \\ 0 & c_{\alpha_i} & -s_{\alpha_i} & 0 \\ 0 & s_{\alpha_i} & c_{\alpha_i} & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

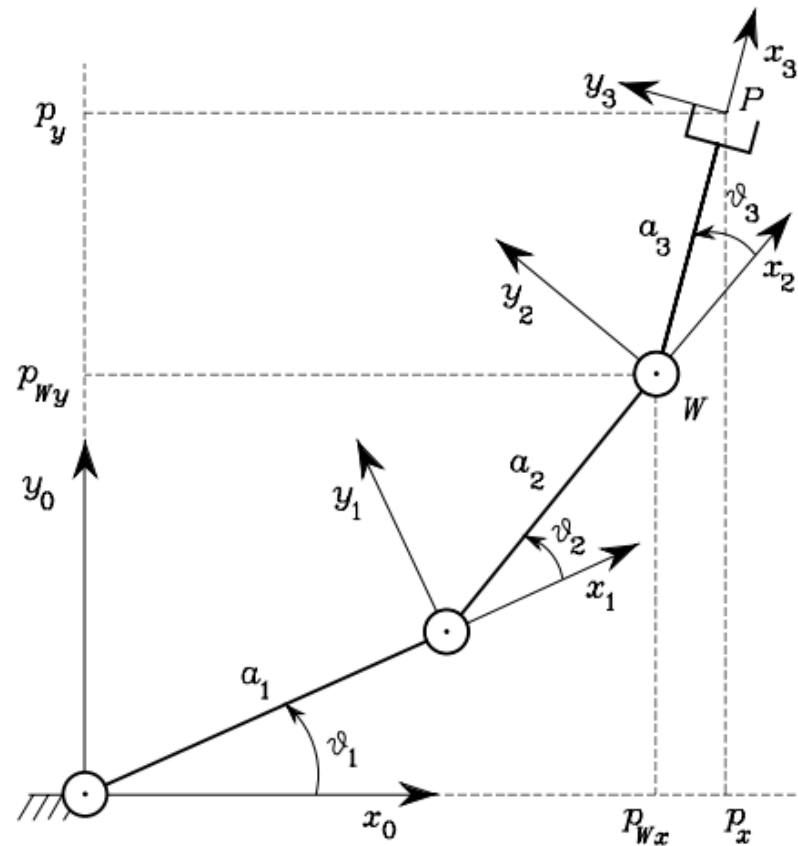
4. Post-multiply the single transformations:

$$\rightarrow \mathbf{A}_i^{i-1}(q_i) = \mathbf{A}_i^{i'} \mathbf{A}_i^{i'} = \begin{bmatrix} c_{\vartheta_i} & -s_{\vartheta_i} c_{\alpha_i} & s_{\vartheta_i} s_{\alpha_i} & a_i c_{\vartheta_i} \\ s_{\vartheta_i} & c_{\vartheta_i} c_{\alpha_i} & -c_{\vartheta_i} s_{\alpha_i} & a_i s_{\vartheta_i} \\ 0 & s_{\alpha_i} & c_{\alpha_i} & d_i \\ 0 & 0 & 0 & 1 \end{bmatrix}$$



2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.1 Three-link Planar Arm



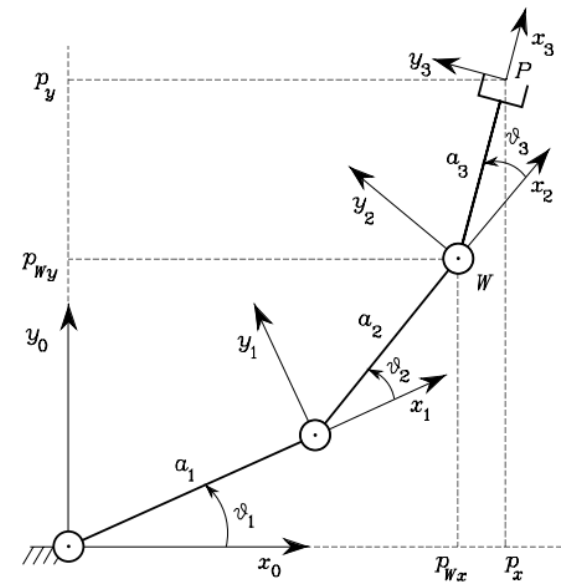
2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.1 Three-link Planar Arm

❖ DH Parameters

Link	a_i	α_i	d_i	ϑ_i
1	a_1	0	0	ϑ_1
2	a_2	0	0	ϑ_2
3	a_3	0	0	ϑ_3

$$\mathbf{A}_i^{i-1}(\vartheta_i) = \begin{bmatrix} c_i & -s_i & 0 & a_i c_i \\ s_i & c_i & 0 & a_i s_i \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \quad i = 1, 2, 3$$



2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.1 Three-link Planar Arm

❖ All joints are revolute:

$$T_3^0(\mathbf{q}) = A_1^0 A_2^1 A_3^2 = \begin{bmatrix} c_{123} & -s_{123} & 0 & a_1 c_1 + a_2 c_{12} + a_3 c_{123} \\ s_{123} & c_{123} & 0 & a_1 s_1 + a_2 s_{12} + a_3 s_{123} \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

$$\mathbf{q} = [\vartheta_1 \quad \vartheta_2 \quad \vartheta_3]^T$$

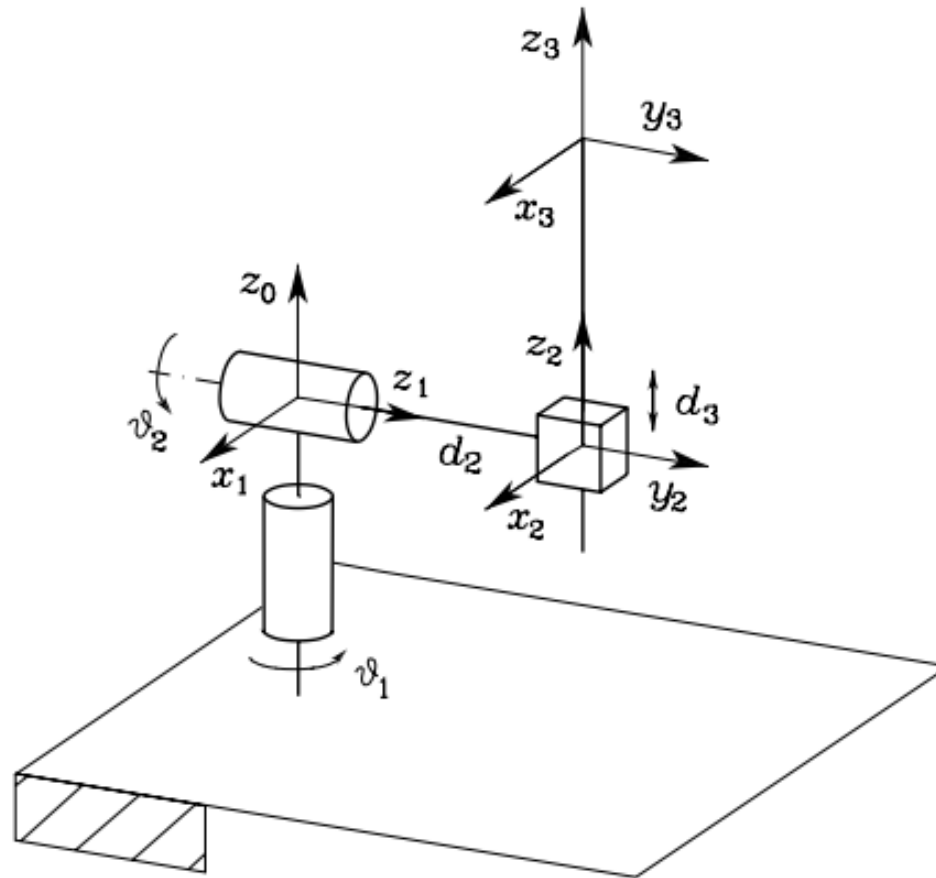
❖ End-effector frame:

$$T_e^3 = \begin{bmatrix} 0 & 0 & 1 & 0 \\ 0 & 1 & 0 & 0 \\ -1 & 0 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$



2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.3 Spherical Arm

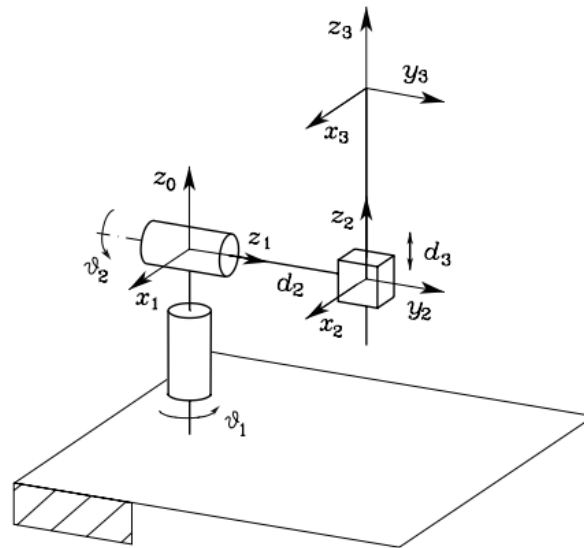


2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.3 Spherical Arm

❖ DH Parameters

Link	a_i	α_i	d_i	ϑ_i
1	0	$-\pi/2$	0	ϑ_1
2	0	$\pi/2$	d_2	ϑ_2
3	0	0	d_3	0



2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.3 Spherical Arm

❖ The homogeneous transformation matrices:

$$\mathbf{A}_1^0(\vartheta_1) = \begin{bmatrix} c_1 & 0 & -s_1 & 0 \\ s_1 & 0 & c_1 & 0 \\ 0 & -1 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \quad \mathbf{A}_2^1(\vartheta_2) = \begin{bmatrix} c_2 & 0 & s_2 & 0 \\ s_2 & 0 & -c_2 & 0 \\ 0 & 1 & 0 & d_2 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

$$\mathbf{A}_3^2(d_3) = \begin{bmatrix} 1 & 0 & 0 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 1 & d_3 \\ 0 & 0 & 0 & 1 \end{bmatrix}.$$

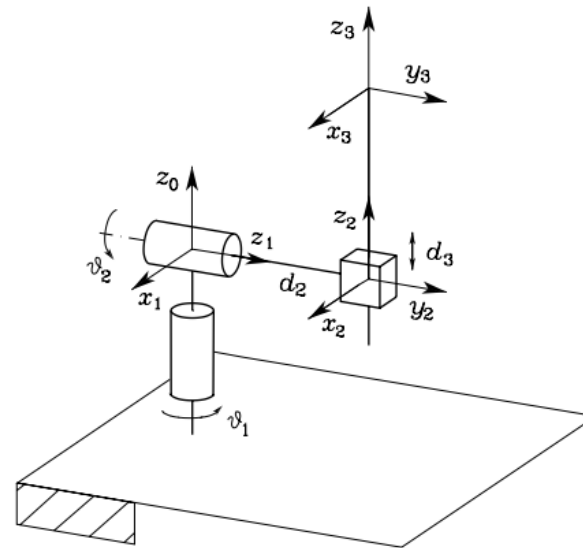
2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.3 Spherical Arm

❖ The direct kinematics function:

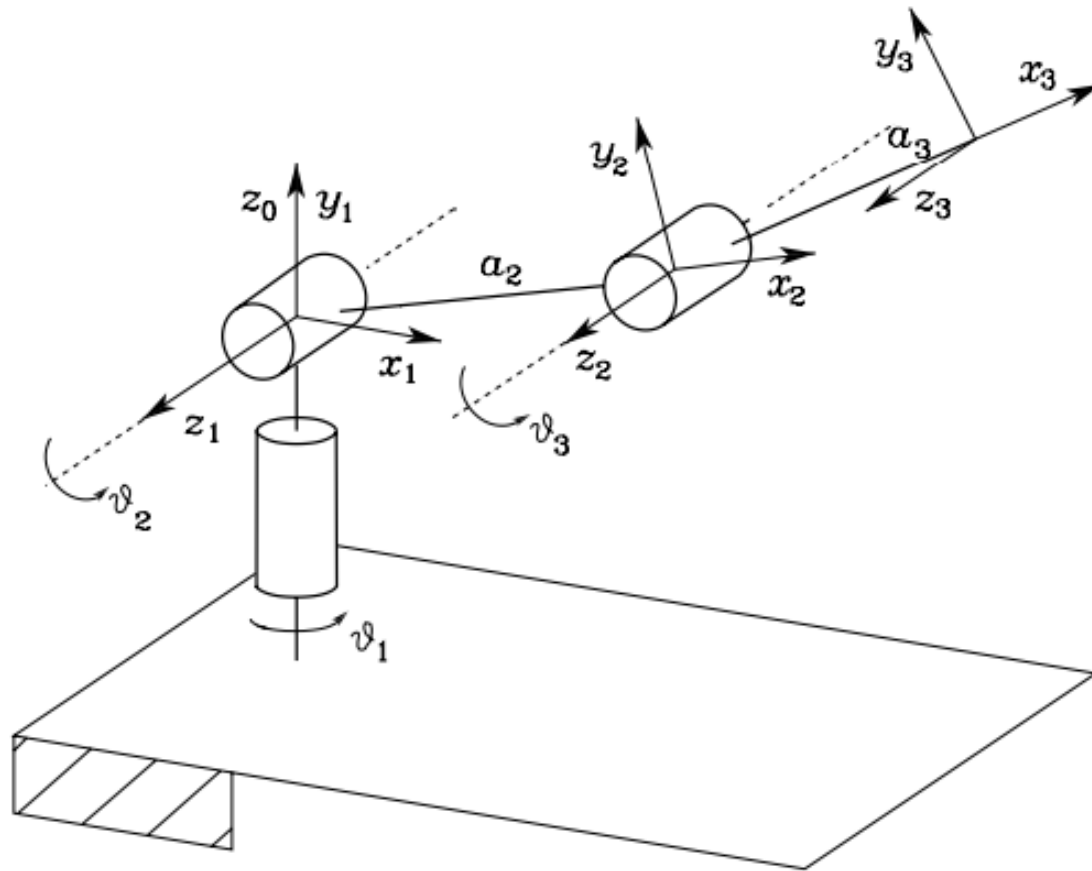
$$\rightarrow T_3^0(\mathbf{q}) = A_1^0 A_2^1 A_3^2 = \begin{bmatrix} c_1 c_2 & -s_1 & c_1 s_2 & c_1 s_2 d_3 - s_1 d_2 \\ s_1 c_2 & c_1 & s_1 s_2 & s_1 s_2 d_3 + c_1 d_2 \\ -s_2 & 0 & c_2 & c_2 d_3 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

$$\mathbf{q} = [\vartheta_1 \quad \vartheta_2 \quad d_3]^T$$



2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.4 Anthropomorphic Arm

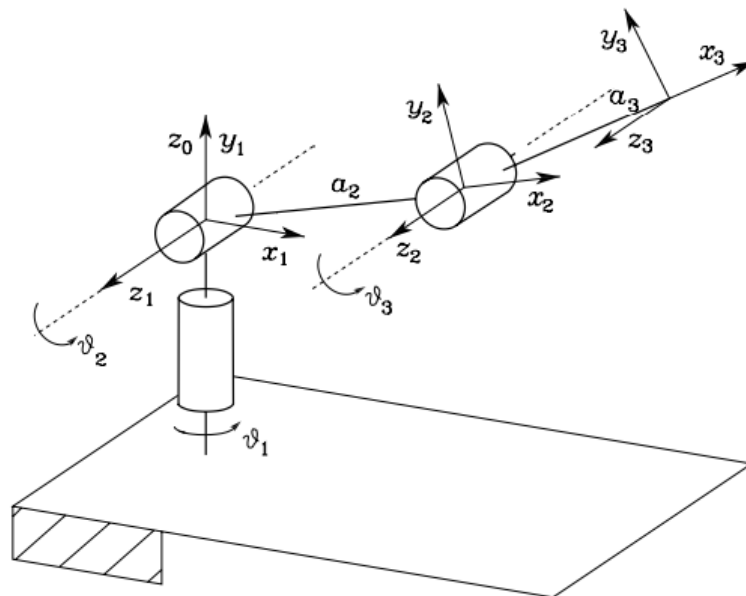


2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.4 Anthropomorphic Arm

❖ DH Parameters:

Link	a_i	α_i	d_i	ϑ_i
1	0	$\pi/2$	0	ϑ_1
2	a_2	0	0	ϑ_2
3	a_3	0	0	ϑ_3



2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

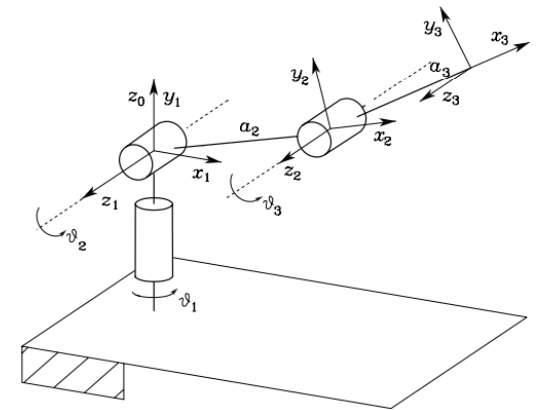
□ 2.9.4 Anthropomorphic Arm

$$\mathbf{A}_1^0(\vartheta_1) = \begin{bmatrix} c_1 & 0 & s_1 & 0 \\ s_1 & 0 & -c_1 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

$$\mathbf{A}_i^{i-1}(\vartheta_i) = \begin{bmatrix} c_i & -s_i & 0 & a_i c_i \\ s_i & c_i & 0 & a_i s_i \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \quad i = 2, 3$$

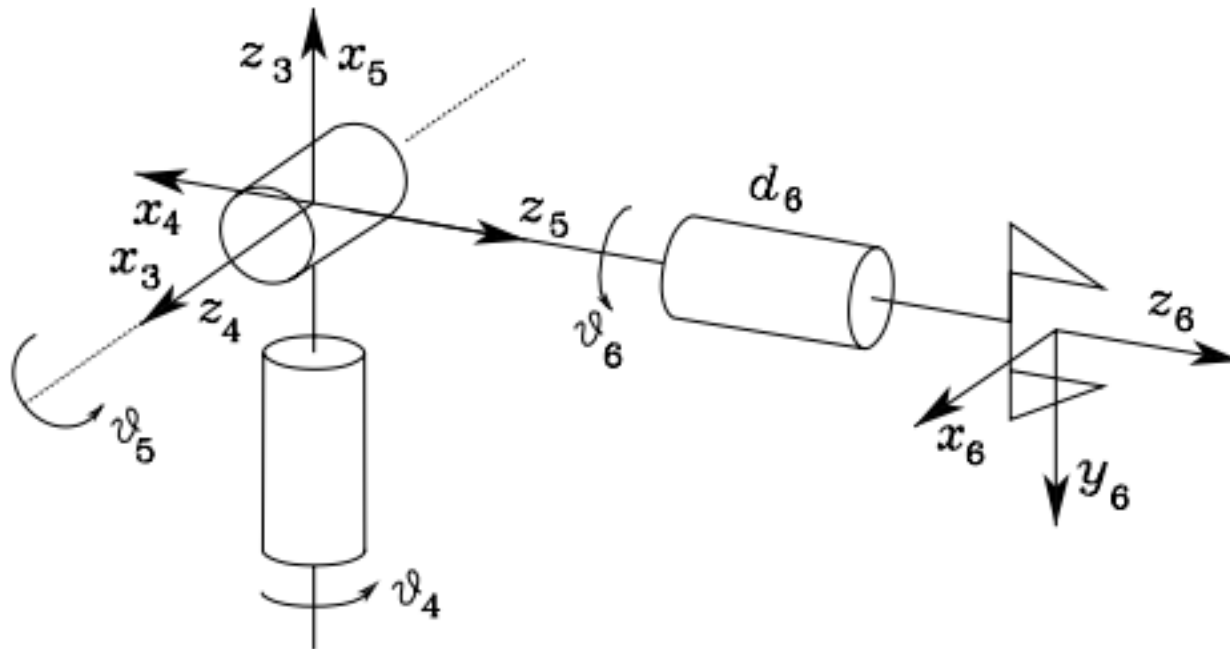
$$\rightarrow \mathbf{T}_3^0(\mathbf{q}) = \mathbf{A}_1^0 \mathbf{A}_2^1 \mathbf{A}_3^2 = \begin{bmatrix} c_1 c_{23} & -c_1 s_{23} & s_1 & c_1 (a_2 c_2 + a_3 c_{23}) \\ s_1 c_{23} & -s_1 s_{23} & -c_1 & s_1 (a_2 c_2 + a_3 c_{23}) \\ s_{23} & c_{23} & 0 & a_2 s_2 + a_3 s_{23} \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

$$\mathbf{q} = [\vartheta_1 \quad \vartheta_2 \quad \vartheta_3]^T$$



2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.5 Spherical Wrist

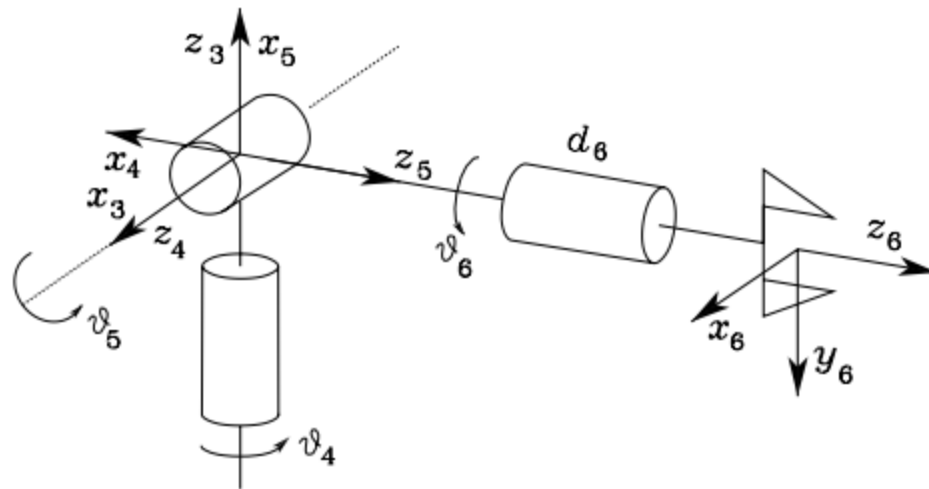


2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.5 Spherical Wrist

❖ DH Parameters:

Link	a_i	α_i	d_i	ϑ_i
4	0	$-\pi/2$	0	ϑ_4
5	0	$\pi/2$	0	ϑ_5
6	0	0	d_6	ϑ_6

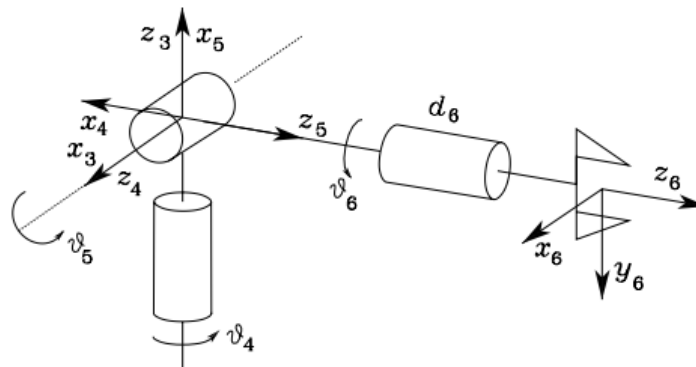


2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.5 Spherical Wrist

$$A_4^3(\vartheta_4) = \begin{bmatrix} c_4 & 0 & -s_4 & 0 \\ s_4 & 0 & c_4 & 0 \\ 0 & -1 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix} \quad A_5^4(\vartheta_5) = \begin{bmatrix} c_5 & 0 & s_5 & 0 \\ s_5 & 0 & -c_5 & 0 \\ 0 & 1 & 0 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

$$A_6^5(\vartheta_6) = \begin{bmatrix} c_6 & -s_6 & 0 & 0 \\ s_6 & c_6 & 0 & 0 \\ 0 & 0 & 1 & d_6 \\ 0 & 0 & 0 & 1 \end{bmatrix}.$$

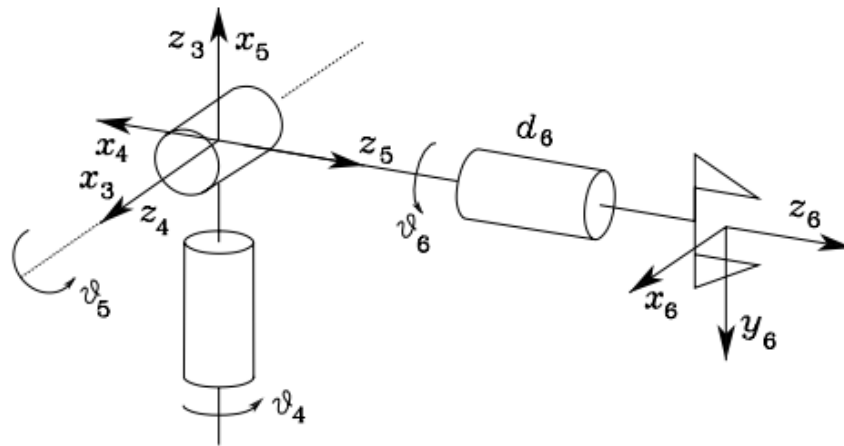


2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.5 Spherical Wrist

$$\rightarrow T_6^3(q) = A_4^3 A_5^4 A_6^5 = \begin{bmatrix} c_4 c_5 c_6 - s_4 s_6 & -c_4 c_5 s_6 - s_4 c_6 & c_4 s_5 & c_4 s_5 d_6 \\ s_4 c_5 c_6 + c_4 s_6 & -s_4 c_5 s_6 + c_4 c_6 & s_4 s_5 & s_4 s_5 d_6 \\ -s_5 c_6 & s_5 s_6 & c_5 & c_5 d_6 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$

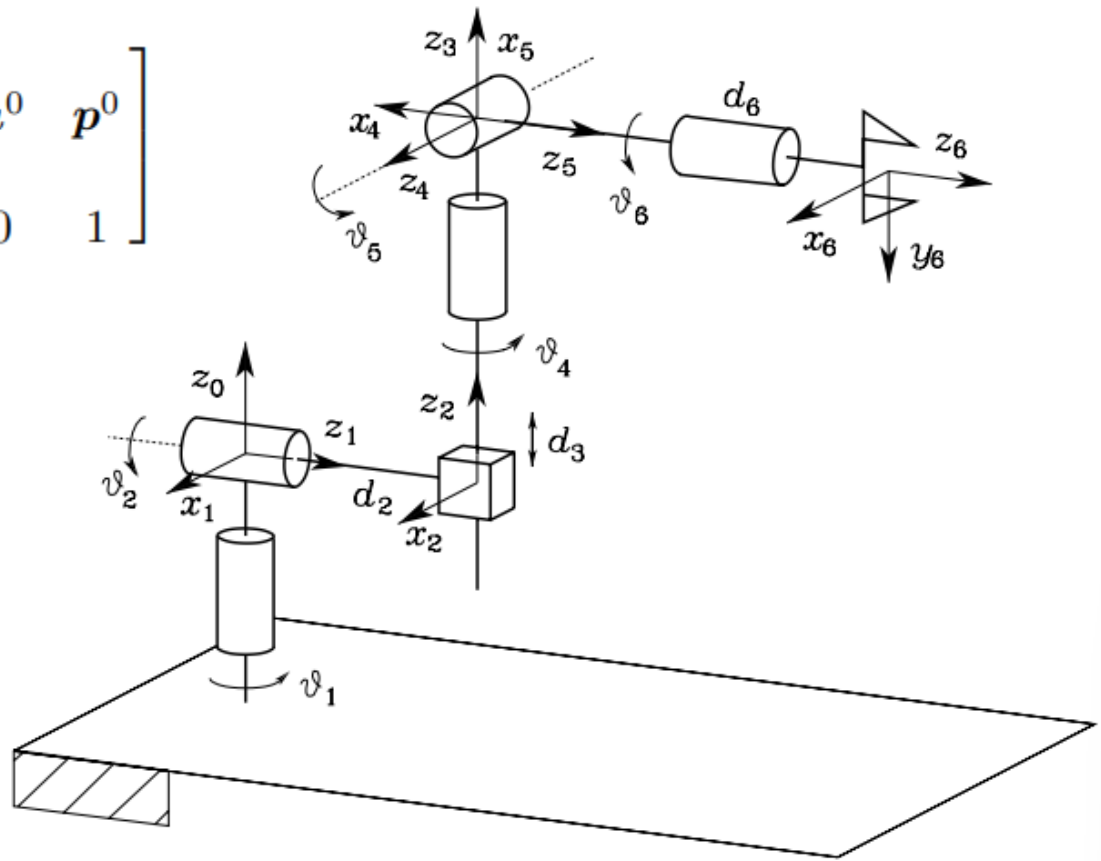
$$q = [\vartheta_4 \quad \vartheta_5 \quad \vartheta_6]^T$$



2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.6 Stanford Manipulator

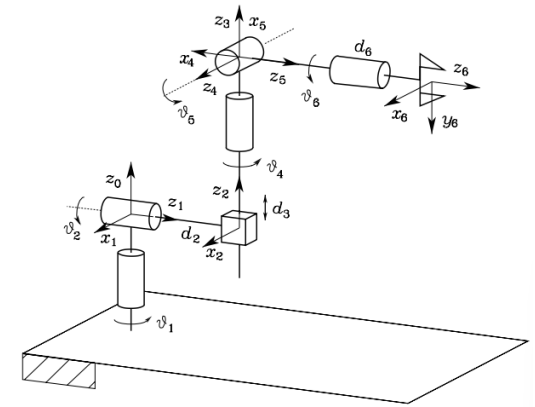
$$T_6^0 = T_3^0 T_6^3 = \begin{bmatrix} n^0 & s^0 & a^0 & p^0 \\ 0 & 0 & 0 & 1 \end{bmatrix}$$



2.9 KINEMATICS OF TYPICAL MANIPULATOR STRUCTURES

□ 2.9.6 Stanford Manipulator

$$\begin{aligned}
 \mathbf{p}_6^0 &= \begin{bmatrix} c_1 s_2 d_3 - s_1 d_2 + (c_1(c_2 c_4 s_5 + s_2 c_5) - s_1 s_4 s_5) d_6 \\ s_1 s_2 d_3 + c_1 d_2 + (s_1(c_2 c_4 s_5 + s_2 c_5) + c_1 s_4 s_5) d_6 \\ c_2 d_3 + (-s_2 c_4 s_5 + c_2 c_5) d_6 \end{bmatrix} \\
 \mathbf{n}_6^0 &= \begin{bmatrix} c_1(c_2(c_4 c_5 c_6 - s_4 s_6) - s_2 s_5 c_6) - s_1(s_4 c_5 c_6 + c_4 s_6) \\ s_1(c_2(c_4 c_5 c_6 - s_4 s_6) - s_2 s_5 c_6) + c_1(s_4 c_5 c_6 + c_4 s_6) \\ -s_2(c_4 c_5 c_6 - s_4 s_6) - c_2 s_5 c_6 \end{bmatrix} \\
 \mathbf{s}_6^0 &= \begin{bmatrix} c_1(-c_2(c_4 c_5 s_6 + s_4 c_6) + s_2 s_5 s_6) - s_1(-s_4 c_5 s_6 + c_4 c_6) \\ s_1(-c_2(c_4 c_5 s_6 + s_4 c_6) + s_2 s_5 s_6) + c_1(-s_4 c_5 s_6 + c_4 c_6) \\ s_2(c_4 c_5 s_6 + s_4 c_6) + c_2 s_5 s_6 \end{bmatrix} \\
 \mathbf{a}_6^0 &= \begin{bmatrix} c_1(c_2 c_4 s_5 + s_2 c_5) - s_1 s_4 s_5 \\ s_1(c_2 c_4 s_5 + s_2 c_5) + c_1 s_4 s_5 \\ -s_2 c_4 s_5 + c_2 c_5 \end{bmatrix}
 \end{aligned}$$



2.10 JOINT SPACE AND OPERATIONAL SPACE

□ Direct Kinematics:

- ❖ Position and orientation of the end-effector frame to be expressed as a function of the joint variables with respect to the base frame.
- ❖ This is quite easy for the position, but quite difficult for orientation (9 components must be guaranteed to satisfy the orthonormality constraints)

- The end-effector pose can be given by a minimal number of coordinates and minimal representation (Euler angles) describing the rotation

$$\mathbf{x}_e = \begin{bmatrix} \mathbf{p}_e \\ \phi_e \end{bmatrix}$$

- ❖ \mathbf{p}_e : End-effector position
- ❖ ϕ_e : End-effector orientation



2.10 JOINT SPACE AND OPERATIONAL SPACE

- The vector x_e is defined in the space in which the manipulator task is specified; hence, this space is typically called *operational space*.

$$x_e = \begin{bmatrix} p_e \\ \phi_e \end{bmatrix}$$

- On the other hand, the *joint space* (configuration space) denotes the space in which the $(n \times 1)$ vector of joint variables

$$q = \begin{bmatrix} q_1 \\ \vdots \\ q_n \end{bmatrix}$$

- ❖ For a revolute joint: $q_i = \vartheta_i$
- ❖ For a prismatic joint: $q_i = d_i$

- *Direct Kinematics Equation:* $x_e = k(q)$

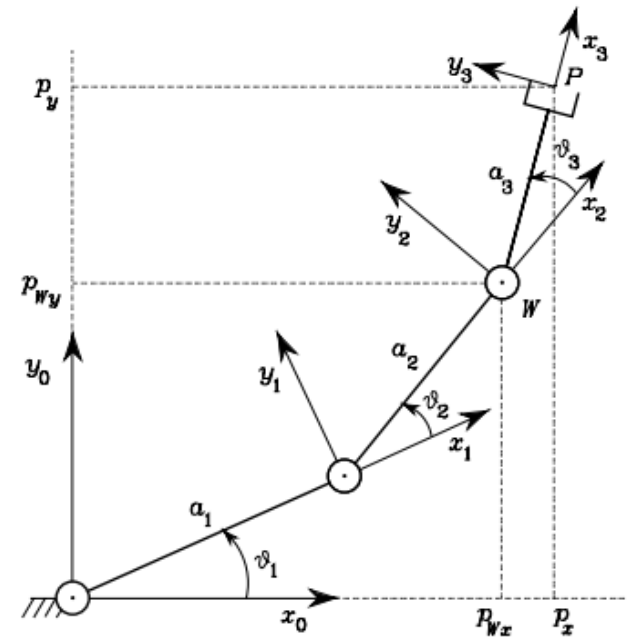


2.10 JOINT SPACE AND OPERATIONAL SPACE

□ Example 2.5

$$\mathbf{x}_e = \begin{bmatrix} p_x \\ p_y \\ \phi \end{bmatrix} = \mathbf{k}(\mathbf{q}) = \begin{bmatrix} a_1 c_1 + a_2 c_{12} + a_3 c_{123} \\ a_1 s_1 + a_2 s_{12} + a_3 s_{123} \\ \vartheta_1 + \vartheta_2 + \vartheta_3 \end{bmatrix}$$

- ❖ 3 joint space variables allow specification of at most 3 independent operational space variables.
- ❖ If orientation is of no concern, there is *kinematic redundancy*.



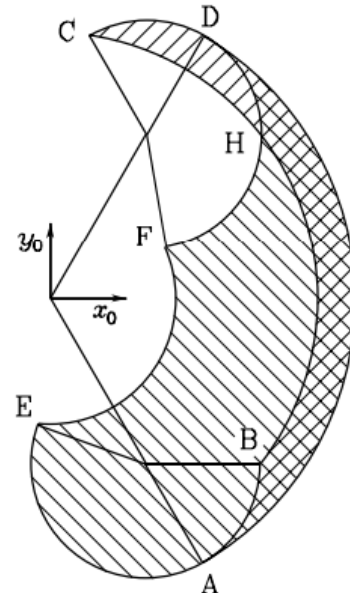
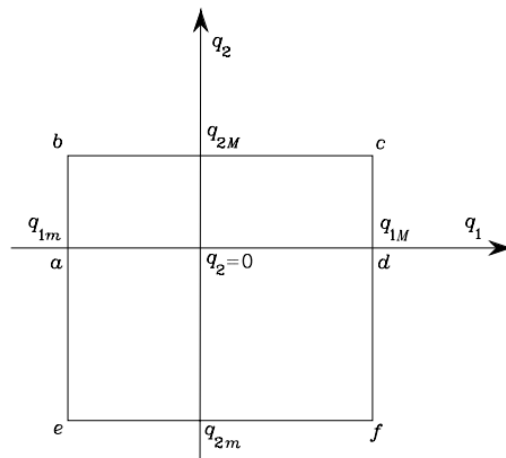
2.10 JOINT SPACE AND OPERATIONAL SPACE

□ 2.10.1 Workspace

- ❖ The region described by the origin of the end-effector frame when all the manipulator joints execute all possible motions
- ❖ This volume is finite, closed, connected and is defined by its bordering surface

□ Example 2.6

- ❖ The simple two-link planar arm



2.10 JOINT SPACE AND OPERATIONAL SPACE

- ❑ 2.10.2 Kinematic Redundancy
- ❑ Kinematically Redundant:
 - ❖ When number of DOFs is greater than the number of variables that are necessary to describe a given task
- ❑ A manipulator is intrinsically redundant when the dimension of the operational space is smaller than the dimension of the joint space ($m < n$)
- ❑ Redundancy is a concept relative to the task assigned to the manipulator.

2.12 INVERSE KINEMATICS PROBLEM

- ❑ The inverse kinematics problem consists of the determination of the joint variables corresponding to a given end-effector position and orientation.
- ❑ It transforms the motion specifications, assigned to the end-effector in the operational space, into the corresponding joint space motions that allow execution of the desired motion.
- ❑ The inverse kinematics problem is much more complex:
 - ❖ The equations to solve are in general nonlinear
 - ❖ Multiple solutions may exist
 - ❖ Infinite solutions may exist
 - ❖ There might be no admissible solutions

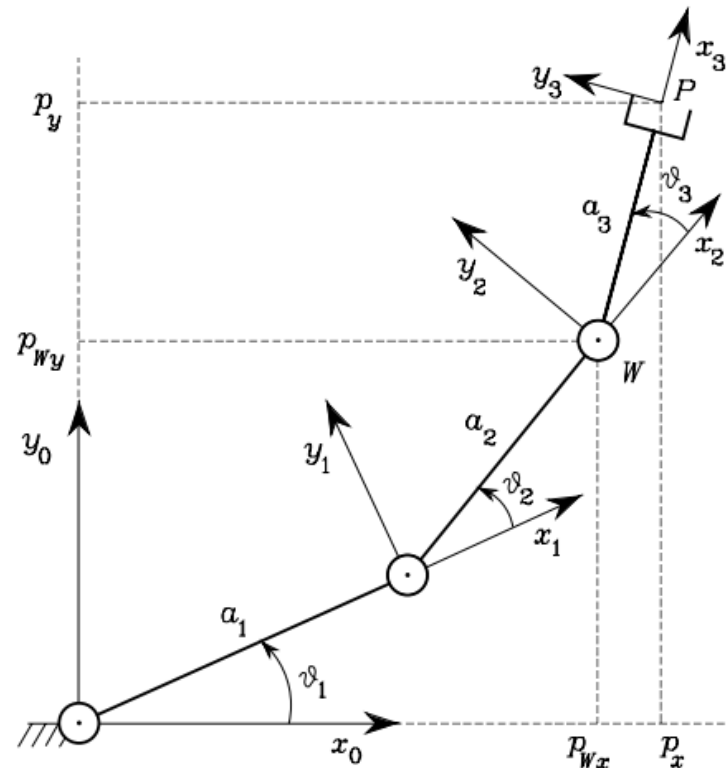


2.12 INVERSE KINEMATICS PROBLEM

□ 2.12.1 Solution of Three-link Planar Arm

❖ The end-effector position and orientation in terms of a minimal number of parameters:

- ✓ The two coordinates p_x, p_y
- ✓ The angle φ with axis x_0



2.12 INVERSE KINEMATICS PROBLEM

□ 2.12.1 Solution of Three-link Planar Arm

❖ *Algebraic solution technique*

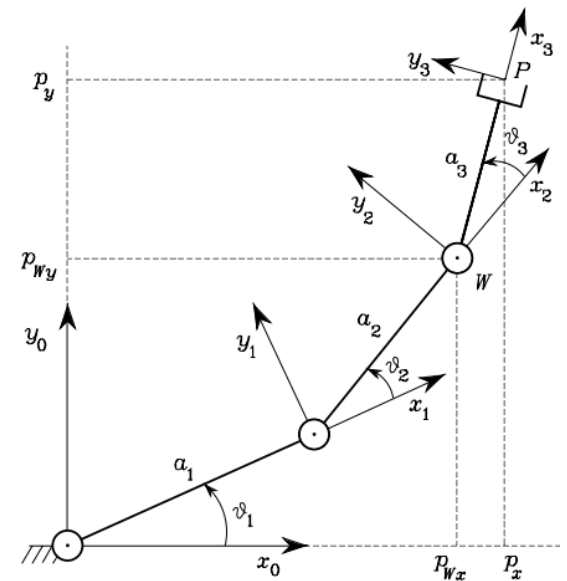
$$\phi = \vartheta_1 + \vartheta_2 + \vartheta_3$$

$$p_{Wx} = p_x - a_3 c_\phi = a_1 c_1 + a_2 c_{12}$$

$$p_{Wy} = p_y - a_3 s_\phi = a_1 s_1 + a_2 s_{12}$$

$$p_{Wx}^2 + p_{Wy}^2 = a_1^2 + a_2^2 + 2a_1 a_2 c_2$$

$$c_2 = \frac{p_{Wx}^2 + p_{Wy}^2 - a_1^2 - a_2^2}{2a_1 a_2}$$



2.12 INVERSE KINEMATICS PROBLEM

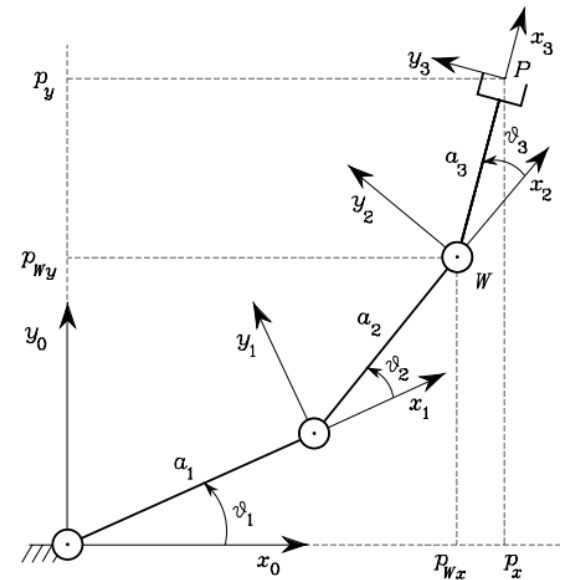
□ 2.12.1 Solution of Three-link Planar Arm

❖ *Algebraic solution technique*

$$c_2 = \frac{p_{Wx}^2 + p_{Wy}^2 - a_1^2 - a_2^2}{2a_1a_2}$$

$$\longrightarrow s_2 = \pm \sqrt{1 - c_2^2}$$

$$\longrightarrow \vartheta_2 = \text{Atan2}(s_2, c_2)$$



2.12 INVERSE KINEMATICS PROBLEM

□ 2.12.1 Solution of Three-link Planar Arm

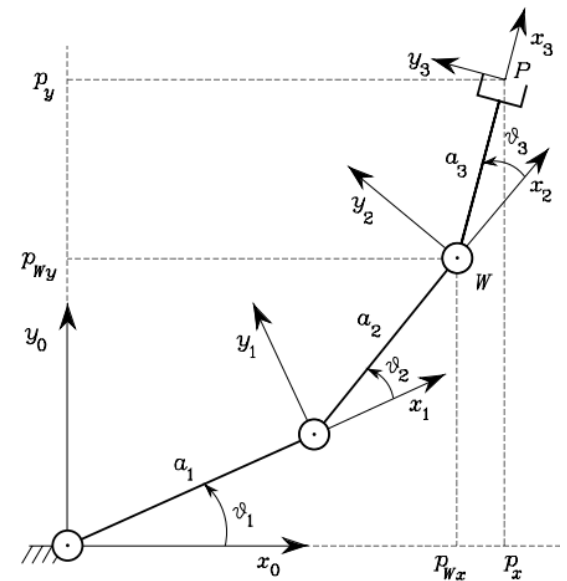
❖ *Algebraic solution technique*

$$s_1 = \frac{(a_1 + a_2 c_2) p_{W_y} - a_2 s_2 p_{W_x}}{p_{W_x}^2 + p_{W_y}^2}$$

$$c_1 = \frac{(a_1 + a_2 c_2) p_{W_x} + a_2 s_2 p_{W_y}}{p_{W_x}^2 + p_{W_y}^2}$$

$$\vartheta_1 = \text{Atan2}(s_1, c_1)$$

$$\vartheta_3 = \phi - \vartheta_1 - \vartheta_2$$



2.12 INVERSE KINEMATICS PROBLEM

□ 2.12.1 Solution of Three-link Planar Arm

❖ *Geometric solution technique*

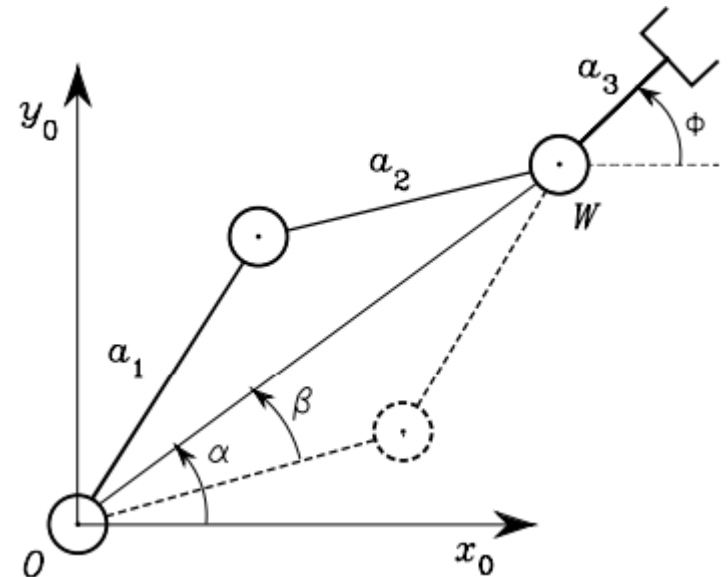
- ✓ The application of the cosine theorem to the triangle formed by links a_1 , a_2 and the segment connecting points W and O

$$p_{Wx}^2 + p_{Wy}^2 = a_1^2 + a_2^2 - 2a_1a_2 \cos(\pi - \vartheta_2)$$

$$\rightarrow c_2 = \frac{p_{Wx}^2 + p_{Wy}^2 - a_1^2 - a_2^2}{2a_1a_2}$$

$$\rightarrow \vartheta_2 = \pm \cos^{-1}(c_2)$$

- ✓ The elbow-up and elbow-down posture



2.12 INVERSE KINEMATICS PROBLEM

□ 2.12.1 Solution of Three-link Planar Arm

❖ *Geometric solution technique*

$$c_\beta \sqrt{p_{Wx}^2 + p_{Wy}^2} = a_1 + a_2 c_2 \quad \rightarrow \quad \beta = \cos^{-1} \left(\frac{p_{Wx}^2 + p_{Wy}^2 + a_1^2 - a_2^2}{2a_1 \sqrt{p_{Wx}^2 + p_{Wy}^2}} \right)$$

$$\alpha = \text{Atan2}(p_{Wy}, p_{Wx})$$

$$\rightarrow \quad \vartheta_1 = \alpha \pm \beta$$

